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## **bubble three-dimensionality in a laminar separation On the origins of unsteadiness and**

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Vassilios Theofilis, Stefan Hein and Uwe Dallmann

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## $\frac{1}{\sqrt{1.5}}$ <br>On the origins of unsteadiness<br>and three-dimensionality in a On the origins of unsteadiness<br>and three-dimensionality in a<br>laminar separation bubble and three-dimensionality in a<br>laminar separation bubble

**laminar separation bubble**<br>By Vassilios Theofilis, Stefan Hein and Uwe Dallmann

*Deutsches Zentrum fur Luft- und Raumfahrt e.V. (DLR), IIIIOS THEOFILIS, STEFAN HEIN AND UWE DALLI*<br>*Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR),*<br>*Institute of Fluid Mechanics, Transition and Turbulence,*<br>*Bamagnatralla 10, D 27072 Cöttingen, Cermany Bunsenstra¼ e 10, D-37073 Gottingen, Germany*

 $\mathcal{L}$  and  $\mathcal{L}$  is the three-dimensional non-parallel instability mechanisms responsible for transition to turbulence in regions of recirculating steady laminar two-dimensional We analyse the three-dimensional non-parallel instability mechanisms responsible for<br>transition to turbulence in regions of recirculating steady laminar two-dimensional<br>incompressible separation bubble flow in a twofold ma transition to turbulence in regions of recirculating steady laminar two-dimensional incompressible separation bubble flow in a twofold manner. First, we revisit the transition to turbulence in regions of recirculating steady laminar two-dimensional<br>incompressible separation bubble flow in a twofold manner. First, we revisit the<br>problem of Tollmien–Schlichting (TS)-like disturbances an incompressible separation bubble flow in a twofold manner. First, we revisit the problem of Tollmien–Schlichting (TS)-like disturbances and we demonstrate, for the first time for this type of flow, excellent agreement betw problem of Tollmien–Schlichting (TS)-like disturbances and we demonstrate, for the<br>first time for this type of flow, excellent agreement between the parabolized stabil-<br>ity equation results and those of independently perfo first time for this type of flow, excellent agreement between the parabolized stability equation results and those of independently performed direct numerical simulations. Second, we perform a partial-derivative eigenvalue ity equation results and those of independently performed direct numerical simulations. Second, we perform a partial-derivative eigenvalue problem stability analysis by discretizing the two spatial directions on which the tions. Second, we perform a partial-derivative eigenvalue problem stability analysis<br>by discretizing the two spatial directions on which the basic flow depends, precluding<br>TS-like waves from entering the calculation domain by discretizing the two spatial directions on which the basic flow depends, precluding TS-like waves from entering the calculation domain. A new two-dimensional set of global amplified instability modes is thus discovered. TS-like waves from entering the calculation domain. A new two-dimensional set of global amplified instability modes is thus discovered. In order to prove earlier topological conjectures about the flow structural changes oc global amplified instability modes is thus discovered. In order to prove earlier topo-<br>logical conjectures about the flow structural changes occurring prior to the onset of<br>bubble unsteadiness, we reconstruct the total flo logical conjectures about the flow structural changes occurring prior to the onset of<br>bubble unsteadiness, we reconstruct the total flowfield by linear superposition of the<br>steady two-dimensional basic flow and the new mos bubble unsteadiness, we reconstruct the total flowfield by linear superposition of the steady two-dimensional basic flow and the new most-amplified global eigenmodes. In the parameter range investigated, the result is a bi steady two-dimensional basic flow and the new most-amplified global eigenmodes. In<br>the parameter range investigated, the result is a bifurcation into a three-dimensional<br>flowfield in which the separation line remains unaff the parameter range investigated, the result is a bifurcation into a three-dimensional flowfield in which the separation line remains unaffected while the primary reattachment line becomes three dimensional, in line with t flowfield in which the separation line remains unaffected while the primary reattachment line becomes three dimensional, in line with the analogous result of a multitude of experimental observations.

> Keywords: linear non-local instability; topological flow changes; laminar separation bubble; global instability; structural instability

### 1. Introduction

1. Introduction<br>The subjects of laminar flow separation, its linear and nonlinear instability and<br>transition to turbulence have attracted experimental and theoretical interest for the The subjects of laminar flow separation, its linear and nonlinear instability and<br>transition to turbulence have attracted experimental and theoretical interest for the<br>heat part of the last century. The understanding of th The subjects of laminar flow separation, its linear and nonlinear instability and<br>transition to turbulence have attracted experimental and theoretical interest for the<br>best part of the last century. The understanding of th transition to turbulence have attracted experimental and theoretical interest for the<br>best part of the last century. The understanding of the physics governing these<br>phenomena, especially with respect to vortex shedding an best part of the last century. The understanding of the physics governing these<br>phenomena, especially with respect to vortex shedding and its relation to instability<br>mechanisms, is still far from being satisfactory, as wit phenomena, especially with respect to vortex shedding and its relation to instability<br>mechanisms, is still far from being satisfactory, as witnessed by the plethora and<br>diversity of contributions to this issue. Motivation mechanisms, is still far from being satisfactory, as witnessed by the plethora and<br>diversity of contributions to this issue. Motivation for continuous study from an<br>aerodynamic point of view is primarily offered by the for diversity of contributions to this issue. Motivation for continuous study from an aerodynamic point of view is primarily offered by the formation of laminar separation bubbles in the neighbourhood of the mid-chord region o aerodynamic point of view is primarily offered by the formation of laminar separation<br>bubbles in the neighbourhood of the mid-chord region of an aerofoil at moderate<br>angles of attack and low Reynolds numbers. Under the inf bubbles in the neighbourhood of the mid-chord region of an aerofoil at moderate angles of attack and low Reynolds numbers. Under the influence of environmental excitation, the laminar separated flow has been demonstrated t angles of attack and low Reynolds numbers. Under the influence of environmental

*Phil. Trans. R. Soc. Lond.* A (2000) 358, 3229-3246

3229

<sup>3230</sup> *V. Theo¯lis,S.HeinandU.Dallmann*

*et al*. 1994) and numerical simulations (see, for example, Bestek *et al*. 1989). As in  $et$  al. 1994) and numerical simulations (see, for example, Bestek  $et$  al. 1989). As in attached boundary-layer flows, essential features of the transition process are the flow unsteadiness and three dimensionality. *et al.* 1994) and numerical simulations (see, inductand boundary-layer flows, essential feat flow unsteadiness and three dimensionality.<br>It should be noted that several parameters It should be noted that several parameters of the transition process are the<br>investigations and three dimensionality.<br>It should be noted that several parameters—such as Reynolds number, means of<br>croduction, extent of appli

flow unsteadiness and three dimensionality.<br>It should be noted that several parameters—such as Reynolds number, means of<br>introduction, extent of application and strength of an adverse pressure gradient—<br>should be taken int It should be noted that several parameters—such as Reynolds number, means of introduction, extent of application and strength of an adverse pressure gradient—should be taken into account in order for a laminar separation b introduction, extent of application and strength of an adverse pressure gradient—<br>should be taken into account in order for a laminar separation bubble to be char-<br>acterized. Rather than relying on parametric studies, howe should be taken into account in order for a laminar separation bubble to be characterized. Rather than relying on parametric studies, however, our concern is with the identification of the physical mechanisms underlying th and three dimensionality of the flow. Specifically, we have the following questions in mind.

- (i) Does the bubble only act as an amplifier of environmental disturbances entering Does the bubble only act as an amplifier of environmental disturbances entering<br>the separated flow region, or does it also have the potential to generate unstable<br>modes in the absence of incoming disturbances? Does the bubble only act as an amplifier of environm<br>the separated flow region, or does it also have the po<br>modes in the absence of incoming disturbances?
- modes in the absence of incoming disturbances?<br>(ii) In either case, what are the structural changes experienced by the laminar separation bubble on account of linear instability mechanisms?
- (iii) Further, how do these structural changes relate to the topological flow changes Further, how do these structural changes relate to the topological flow changes<br>which have been conjectured in the literature as being responsible for the onset<br>of vortex shedding from separation bubbles? Further, how do these structural changes relate<br>which have been conjectured in the literature as<br>of vortex shedding from separation bubbles?

of vortex shedding from separation bubbles?<br>Abandoning any intention of offering a complete discussion here of the large body of<br>work concerning instability and transition of flow encompassing a laminar separation or vortex sheading from separation bubbles.<br>Abandoning any intention of offering a complete discussion here of the large body of<br>work concerning instability and transition of flow encompassing a laminar separation<br>bubble w Abandoning any intention of offering a complete discussion here of the large body of<br>work concerning instability and transition of flow encompassing a laminar separation<br>bubble, we confine ourselves to drawing the demarcat work concerning instability and transition of flow encompassing a laminar separation bubble, we confine ourselves to drawing the demarcation line between the approaches used in the past and those presently utilized in orde We also stress that, rather than performing a study at Reynolds numbers relevant used in the past and those presently utilized in order to address the above questions.<br>We also stress that, rather than performing a study at Reynolds numbers relevant<br>to industrial applications (as done, for example, by We also stress that, rather than performing a study at F<br>to industrial applications (as done, for example, by Spala<br>confine ourselves to low transitional Reynolds numbers.<br>We address the questions posed within the frame of industrial applications (as done, for example, by Spalart & Coleman (1997)), we nfine ourselves to low transitional Reynolds numbers.<br>We address the questions posed within the frame of linear instability analyses which di

confine ourselves to low transitional Reynolds numbers.<br>We address the questions posed within the frame of linear instability analyses<br>in which disturbances are treated as non-periodic two-dimensional functions of the<br>chor We address the questions posed within the frame of linear instability analyses<br>in which disturbances are treated as non-periodic two-dimensional functions of the<br>chordwise and wall-normal spatial coordinates. Two complemen in which disturbances are treated as non-periodic two-dimensional functions of the chordwise and wall-normal spatial coordinates. Two complementary approaches have been followed. First, we consider slow development of the chordwise and wall-normal spatial coordinates. Two complementary approaches have<br>been followed. First, we consider slow development of the basic flow in the streamwise<br>direction and apply our non-local analysis tools, bas been followed. First, we consider slow development of the basic flow in the streamwise<br>direction and apply our non-local analysis tools, based on the parabolized stability<br>equations (PSE) developed by Hein *et al.* (1994). direction and apply our non-local analysis tools, based on the parabolized stability<br>equations (PSE) developed by Hein *et al.* (1994). Second, we perform a partial-<br>derivative eigenvalue problem stability (PEPS) analysis equations (PSE) developed by Hein *et al.* (1994). Second, we perform a partial-<br>derivative eigenvalue problem stability (PEPS) analysis (Theofilis 1998) in which<br>both the chordwise and wall-normal directions are fully re derivative eigenvalue problem stability (PEPS) analysis (Theofilis 1998) in which<br>both the chordwise and wall-normal directions are fully resolved. In both approaches<br>the linear disturbances are taken to be periodic in the both the chordwise and wall-normal directions are fully resolved. In both approaches the linear disturbances are taken to be periodic in the spanwise direction. It should be noted that we study solely instabilities of a convective nature when employing the PSE, since evidence exists (Allen & Riley 1995) that breakdown to turbulence in basic flows with small separation bubbles, in terms of extent of the bubble compared with a typical length and in terms of magnitude of t basic flows with small separation bubbles, in terms of extent of the bubble compared with a typical length and in terms of magnitude of the steady recirculating flow compared with the freestream velocity, is of a convecti with a typical length and in terms of magnitude of the steady recirculating flow

compared with the freestream velocity, is of a convective nature.<br>In  $\S 2$  we present the two basic flows utilized for our numerical linear instability<br>experiments. Elements of the theories governing the non-local and glo In §2 we present the two basic flows utilized for our numerical linear instability<br>experiments. Elements of the theories governing the non-local and global instability<br>analyses based on the PSE and the PEPS, respectively, experiments. Elements of the theories governing the non-local and global instability<br>analyses based on the PSE and the PEPS, respectively, are presented in § 3. In<br>§ 4 we present a comparison of our PSE results with those analyses based on the PSE and the PEPS, respectively, are presented in § 3. In § 4 we present a comparison of our PSE results with those of the spatial direct numerical simulations (DNS) of Rist & Maucher (1994) and class  $\S 4$  we present a comparison of our PSE results with those of the spatial direct<br>numerical simulations (DNS) of Rist & Maucher (1994) and classic Orr-Sommerfeld<br>theory. New global linear instability modes discovered by t numerical simulations (DNS) of Rist & Maucher (1994) and classic Orr–Sommerfeld theory. New global linear instability modes discovered by the PEPS analysis are also presented in  $\S 4$ . Section 5 is devoted to discussing o presented in §4. Section 5 is devoted to discussing our findings from a topological *Phil. Trans. R. Soc. Lond.* A (2000)

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Unsteadiness and three-dimensionality in separation bubbles 3231<br>analysis point of view. Concluding remarks relating our results to earlier work are<br>offered in  $86$ analysis point  $\alpha$ <br>offered in  $\S\,6.$ 

### 2. The basic flows

The instability properties and characteristics of two different incompressible two-The instability properties and characteristics of two different incompressible two-<br>dimensional flat-plate laminar boundary-layer flows with separation bubbles encom-<br>passed are analysed. In both cases laminar separation i The instability properties and characteristics of two different incompressible two-<br>dimensional flat-plate laminar boundary-layer flows with separation bubbles encom-<br>passed are analysed. In both cases laminar separation dimensional flat-plate laminar boundary-layer flows with separation bubbles encom-<br>passed are analysed. In both cases laminar separation is caused by a decelerated<br>freestream velocity  $U_e(x)$  prescribed at some fixed dista passed are analysed. In both cases laminar separation is caused by a decelerated<br>freestream velocity  $U_e(x)$  prescribed at some fixed distance from the wall. The insta-<br>bilities developing upon the first basic flow (BF1) w bilities developing upon the first basic flow (BF1) were studied in a DNS by Rist & Maucher (1994). They prescribed a smooth deceleration of  $\Delta U_e = 0.09$  between bilities developing upon the first basic flow (BF1) were studied in a DNS by Rist & Maucher (1994). They prescribed a smooth deceleration of  $\Delta U_e = 0.09$  between  $x_1 = 0.71$  and  $x_2 = 2.43$ , where x and y denote streamwis & Maucher (1994). They prescribed a smooth deceleration of  $\Delta U_e = 0.09$  between  $x_1 = 0.71$  and  $x_2 = 2.43$ , where x and y denote streamwise and wall-normal direction, respectively, and  $(U, V)^T$  denotes the corresponding v tion, respectively, and  $(U, V)^T$  denotes the corresponding velocity vector. Upstream of  $x_1$  and downstream of  $x_2$  the freestream velocity is kept constant. Flow quantities tion, respectively, and  $(U, V)^T$  denotes the corresponding velocity vector. Upstream<br>of  $x_1$  and downstream of  $x_2$  the freestream velocity is kept constant. Flow quantities<br>are made non-dimensional with the freestream v of  $x_1$  and downstream of  $x_2$  the freestream velocity is kept constant. Flow quantities<br>are made non-dimensional with the freestream velocity  $U_{\infty}^* = 30 \text{ m s}^{-1}$  upstream<br>of  $x_1$  and a global reference length  $L^*$  $Re = U_{\infty}^* L^* / \nu^*$ on-dimensional with the freestream velocity  $U_{\infty}^{*} = 30 \text{ m s}^{-1}$  upstream<br>a global reference length  $L^{*} = 0.05 \text{ m}$ . The global Reynolds number<br> $\frac{*}{\nu^{*}}$  was set to  $Re = 10^{5}$ . The steady basic flow was calculated of  $x_1$  and a global reference length  $L^* = 0.05$  m. The global Reynolds number<br>  $Re = U^*_{\infty} L^*/\nu^*$  was set to  $Re = 10^5$ . The steady basic flow was calculated by<br>
Rist & Maucher (1994) using their DNS code with a Blasiu Rist & Maucher (1994) using their DNS code with a Blasius solution as the inflow condition prescribed at  $x_0 = 0.37$ . The flow separates at  $x_S \approx 1.73$  and reattaches at Rist & Maucher (1994) using their DNS code with a Blasius solution as the inflow<br>condition prescribed at  $x_0 = 0.37$ . The flow separates at  $x_S \approx 1.73$  and reattaches at<br> $x_R \approx 2.21$ . Downstream of the bubble the flow asym condition prescribed at  $x_0 = 0.37$ . The flow separates at  $x_S \approx 1.73$  and rea  $x_R \approx 2.21$ . Downstream of the bubble the flow asymptotically recovers to boundary layer, as indicated by the shape factor  $H_{12}$  (Hein *et a*  $\chi$   $\approx$  2.21. Downstream of the bubble the flow asymptotically recovers to a Blasius<br>undary layer, as indicated by the shape factor  $H_{12}$  (Hein *et al.* 1998).<br>Linear instability mechanisms supported by a second basic

boundary layer, as indicated by the shape factor  $H_{12}$  (Hein *et al.* 1998).<br>Linear instability mechanisms supported by a second basic flow (BF2), that described by Briley (1971), have also been studied. This flow is ca Linear instability mechanisms supported by a second basic flow (BF2), that described by Briley (1971), have also been studied. This flow is calculated in two stages. First, the non-similar boundary-layer equations are solv scribed by Briley (1971), have also been studied. This flow is calculated in two<br>stages. First, the non-similar boundary-layer equations are solved subject to the clas-<br>sic linearly decelerating streamwise velocity compone stages. First, the non-similar boundary-layer equations are solved subject to the classic linearly decelerating streamwise velocity component distribution due to Howarth (1938). Second, a two-dimensional incompressible DNS sic linearly decelerating streamwise velocity component distribution due to Howarth (1938). Second, a two-dimensional incompressible DNS is performed using an efficient and spectrally accurate algorithm based on matrix dia (1938). Second, a two-dimensional incompressible DNS is performed using an efficient and spectrally accurate algorithm based on matrix diagonalization. The inflow boundary data for the DNS are provided by the boundary-laye cient and spectrally accurate algorithm based on matrix diagonalization. The inflow<br>boundary data for the DNS are provided by the boundary-layer solution taken at<br>a location well before separation occurs. The imposition in boundary data for the DNS are provided by the boundary-layer solution taken at a location well before separation occurs. The imposition in the DNS of the same freestream velocity distribution as in Howarth's (1938) case up a location well before separation occurs. The imposition in the DNS of the same<br>freestream velocity distribution as in Howarth's (1938) case up to a given down-<br>stream position, beyond which the freestream velocity is kept freestream velocity distribution as in Howarth's (1938) case up to a given down-<br>stream position, beyond which the freestream velocity is kept constant, results in<br>the formation of a separated flow region which compares v the formation of a separated flow region which compares very well with those presented by both Briley (1971) and Cebeci & Stewartson (1983). The parameters held the formation of a separated flow region which compares very well with those pre-<br>sented by both Briley (1971) and Cebeci & Stewartson (1983). The parameters held<br>constant were the Reynolds number,  $Re = 10^6/48$ , and the i sented by both Briley (1971) and Cebeci & Stewartson (1983). The parameters held<br>constant were the Reynolds number,  $Re = 10^6/48$ , and the inflow location,  $\xi_{\text{inf}} = 0.05$ <br>(corresponding to  $x_{\text{inf}} = (\beta_0/\beta_1)\xi_{\text{inf}} = 0.05$ (corresponding to  $x_{\text{inf}} = (\beta_0/\beta_1)\xi_{\text{inf}} = 0.05/3$ ), while those varied were the down-(corresponding to  $x_{\text{inf}} = (\beta_0/\beta_1)\xi_{\text{inf}} = 0.05/3$ ), while those varied were the down-<br>stream extent of the calculation domain, taking values from  $\xi = 0.5$  (Briley 1971)<br>to  $\xi = 1.5$  (Cebeci & Stewartson 1983), and the stream extent of the calculation domain, taking values from  $\xi = 0.5$  (Briley 1971)<br>to  $\xi = 1.5$  (Cebeci & Stewartson 1983), and the (scaled) location in the wall-normal<br>direction where the pressure gradient is imposed, v to  $\xi = 1.5$  (Cebeci & Stewartson 1983), and the (scaled) location in the wall-normal direction where the pressure gradient is imposed, varying from  $\eta = 5.4$  in the former reference to  $\eta = 7.5$  in the latter reference. rection where the pressure gradient is imposed, varying from  $\eta = 5.4$  in the former<br>ference to  $\eta = 7.5$  in the latter reference.<br>One essential difference between BF1 and BF2 is that, in the former basic flow,<br>e Beynolds

reference to  $\eta = 7.5$  in the latter reference.<br>One essential difference between BF1 and BF2 is that, in the former basic flow,<br>the Reynolds number based on the displacement thickness  $\delta_1$  at  $x_1 = 0.71$  is  $Re_{\delta_1} \approx$ <br> the Reynolds number based on the displacement thickness  $\delta_1$  at  $x_1 = 0.71$  is  $Re_{\delta_1} \approx 460$ . According to linear local instability theory, the critical Reynolds number of the Reynolds number based on the displacement thickness  $\delta_1$  at  $x_1 = 0.71$  is  $Re_{\delta_1} \approx 460$ . According to linear local instability theory, the critical Reynolds number of Blasius flow is  $Re_{\delta_1} = 420$  (see, for exam 460. According to linear local instability theory, the critical Reynolds number of Blasius flow is  $Re_{\delta_1} = 420$  (see, for example, Schlichting 1979). Hence, for this basic flow, Tollmien–Schlichting instability sets in Blasius flow is  $Re_{\delta_1} = 420$  (see, for example, Schlichting 1979). Hence, for this basic<br>flow, Tollmien–Schlichting instability sets in upstream of the position  $x_1$ , where flow<br>deceleration starts. In BF2, on the othe flow, Tollmien–Schlichting instability sets in upstream of the position  $x_1$ , where flow<br>deceleration starts. In BF2, on the other hand, the Reynolds number at the outflow<br>boundary  $\xi_{\text{out}} = 1.5$ , where a Blasius bounda deceleration starts. In BF2, on the other has<br>boundary  $\xi_{\text{out}} = 1.5$ , where a Blasius boundary  $\xi_{\text{out}} = 1.5$ , where a Blasius critical  $Re_{\delta_1}$ . well upstream of the Blasius critical  $Re_{\delta_1}$ .

*Phil. Trans. R. Soc. Lond.* A (2000)

<sup>3232</sup> *V. Theo¯lis,S.HeinandU.Dallmann* Theofilis, S. Hein and U. Dallmann<br>3. Linear instability theories

 $\frac{3.5}{3}$  S. Linear instability theories<br>The non-local/global nature of the instability analyses used derives from the decom-<br>position of a perturbed flowfield into a steady two-dimensional basic and an unsteady The non-local/global nature of the instability analyses used derives from the decom-<br>position of a perturbed flowfield into a steady two-dimensional basic and an unsteady<br>three-dimensional disturbance component according t The non-local/global nature of the instability analyses use<br>position of a perturbed flowfield into a steady two-dimensio<br>three-dimensional disturbance component according to position of a perturbed flowfield into a steady two-dimensional basic and an unsteady three-dimensional disturbance component according to

$$
\mathbf{Q}(x, y, z, t) = \mathbf{Q}_{\mathrm{b}}(x, y) + \varepsilon \mathbf{Q}_{\mathrm{p}}(x, y) \exp i\Theta + \mathrm{c.c.},\tag{3.1}
$$

 $\mathbf{Q}(x, y, z, t) = \mathbf{Q}_{\text{b}}(x, y) + \varepsilon \mathbf{Q}_{\text{p}}(x, y) \exp{i\Theta} + \text{c.c.},$  (3.1)<br>with  $\mathbf{Q}_{\text{b}} = (U, V, 0, P)^{\text{T}}$  indicating the real basic flow and  $\mathbf{Q}_{\text{p}} = (\hat{u}, \hat{v}, \hat{w}, \hat{p})^{\text{T}}$  denoting<br>complex-valued perturbations with  $\mathbf{Q}_{\rm b} = (U, V, 0, P)^{\rm T}$  indicating the real basic flow and  $\mathbf{Q}_{\rm p} = (\hat{u}, \hat{v}, \hat{w}, \hat{p})^{\rm T}$  denoting<br>complex-valued perturbations to it. Here, x denotes the streamwise direction, y the<br>wall-normal direction with  $\mathbf{Q}_{\rm b} = (U, V, 0, P)^{\rm T}$  indicating the real basic flow and  $\mathbf{Q}_{\rm p} = (\hat{u}, \hat{v}, \hat{w}, \hat{p})^{\rm T}$  denoting<br>complex-valued perturbations to it. Here, x denotes the streamwise direction, y the<br>wall-normal direction complex-valued perturbations to it. Here, x denotes the streamwise direction, y the wall-normal direction and z the spanwise spatial direction; the imaginary unit is  $i = \sqrt{-1}$  and c.c. denotes complex conjugation in order wall-normal direction and z the spanwise spatial direction; the imaginary unit is  $i = \sqrt{-1}$  and c.c. denotes complex conjugation in order for the total field to remain real. Linearization about  $Q_b$  follows, based on the  $i = \sqrt{-1}$  and c.c. denotes complex conjugation in order for the total field to remain<br>real. Linearization about  $Q_{\rm b}$  follows, based on the argument of smallness of  $\varepsilon$  and the<br>basic flow terms, themselves satisfying real. Linearization about  $Q_b$  follows, based on the argument of smallness of  $\varepsilon$  and the basic flow terms, themselves satisfying the Navier-Stokes and continuity equations, are subtracted out. The difference between th basic flow terms, themselves satisfying the Navare subtracted out. The difference between the  $\mathfrak t$  made for the complex function  $\Theta$ , as follows. made for the complex function  $\Theta$ , as follows.<br>(a) *Non-local analysis based on the PSE* 

The linear non-local instability analyses were performed with the NOnLOcal Tran-The linear non-local instability analyses were performed with the NOnLOcal Transition analysis (NOLOT)/PSE code developed in cooperation between DLR and Fly-<br>
stekniska Forsokanstalten (Hein et al. 1994)  $\dagger$  In the NOLOT The linear non-local instability analyses were performed with the NOnLOcal Transition analysis (NOLOT)/PSE code developed in cooperation between DLR and Flygtekniska Forsokanstalten (Hein *et al.* 1994).<sup>†</sup> In the NOLOT/P gtekniska Forsokanstalten (Hein *et al.* 1994).<sup>†</sup> In the NOLOT/PSE analysis, both  $Q_{\text{b}}$  and  $Q_{\text{p}}$  are functions of y slowly varying in x; we call this approach a *non-local* gtekniska Forsokanstalten (Hein *et al.* 1994).<sup>†</sup> In the NOLOT/PSE analysis, both  $Q_b$  and  $Q_p$  are functions of *y* slowly varying in *x*; we call this approach a *non-local* analysis, as opposed to the classic *local*  $Q_{\rm b}$  and  $Q_{\rm p}$  are functions of y sloven<br>analysis, as opposed to the classic<br>of y alone. The wave function

on  
\n
$$
\Theta = \int_{x_0}^{x} \alpha(\xi) d\xi + \beta z - \Omega t, \qquad (3.2)
$$

with  $\alpha$  a slowly varying streamwise wavenumber, is intended to capture practically with  $\alpha$  a slowly varying streamwise wavenumber, is intended to capture practically<br>all oscillatory-type streamwise variations of the disturbance, that is the variations on<br>the fast scale. The disturbances are assumed to with  $\alpha$  a slowly varying streamwise wavenumber, is intended to capture practically<br>all oscillatory-type streamwise variations of the disturbance, that is the variations on<br>the fast scale. The disturbances are assumed to all oscillatory-type streamwise variations of the disturbance, that is the variations on<br>the fast scale. The disturbances are assumed to grow in the streamwise direction x<br>only, and spatially evolving wave-like instabilit the fast scale. The disturbances are assumed to grow in the streamwise direction  $x$  only, and spatially evolving wave-like instabilities are considered. Hence, the streamwise wavenumber  $\alpha$  is represented by a complex q wavenumber  $\beta$  and the circular frequency  $\Omega$  are real parameters. Since both the amplitude function and the wave function depend on the x-direc-<br>Since both the amplitude function and the wave function depend on the x-direc-<br>on a normalization condition is required which removes this amb

wavenumber  $\beta$  and the circular frequency  $\Omega$  are real parameters.<br>Since both the amplitude function and the wave function depend on the *x*-direction, a normalization condition is required which removes this ambiguity, Since both the amplitude function, a normalization condition is induced by Bertolotti *et al.* (1992):

):  

$$
\int_0^{y_e} \mathbf{Q}_p^{\dagger} \frac{\partial \mathbf{Q}_p}{\partial x} dy = 0.
$$
 (3.3)

 $\int_0^{\infty} \frac{Q_p}{\partial x} dy = 0.$  (3.3)<br>The superscript <sup>†</sup> refers to the complex conjugate and  $y_e$  stands for the upper bound-<br>ary of the discretized domain. The normalization condition is used as an iteration The superscript  $^{\dagger}$  refers to the complex conjugate and  $y_e$  stands for the upper bound-<br>ary of the discretized domain. The normalization condition is used as an iteration<br>condition in order to ensure that as much as The superscript <sup>†</sup> refers to the complex conjugate and  $y_e$  stands for the upper bound-<br>ary of the discretized domain. The normalization condition is used as an iteration<br>condition in order to ensure that as much as poss ary of the discretized domain. The normalization condition is used as an iteration condition in order to ensure that as much as possible of the streamwise variation of the disturbance is transferred into the wave function. condition in order to ensure that as much as possible of the streamwise variation<br>of the disturbance is transferred into the wave function. The remaining part of the<br>streamwise variations is described by the amplitude fun of the disturbance is transferred into the wave function. The remaining part of the streamwise variations is described by the amplitude functions and is small; consequently, second derivatives of the amplitude functions i

ently, second derivatives of the amplitude functions in  $x$  are discarded. This leads<br>† For the purposes of the present study the code was run in the incompressible limit at which excellent<br>reement with results obtained b agreement with results obtained by incompressible codes has been shown by Hein *et al*. (1994).

**MATHEMATICAL,<br>PHYSICAL**<br>& ENGINEERING<br>SCIENCES

THE ROYAL

*Unsteadiness andthree-dimensionalityin separation bubbles* <sup>3233</sup>

to a system of partial differential equations of the form

differential equations of the form  
\n
$$
A\mathbf{Q}_{\mathrm{p}} + B\frac{\partial \mathbf{Q}_{\mathrm{p}}}{\partial y} + C\frac{\partial^2 \mathbf{Q}_{\mathrm{p}}}{\partial y^2} + D\frac{\partial \mathbf{Q}_{\mathrm{p}}}{\partial x} = 0.
$$
\n(3.4)

Details of the consistent multiple-scales approach based on a Reynolds number Details of the consistent multiple-scales approach based on a Reynolds number<br>expansion used to derive the non-local stability equations, which are closely related<br>to the PSE (Bertolotti *et al.* 1992: Herbert 1997), and Details of the consistent multiple-scales approach based on a Reynolds number<br>expansion used to derive the non-local stability equations, which are closely related<br>to the PSE (Bertolotti *et al.* 1992; Herbert 1997), and t to the PSE (Bertolotti *et al.* 1992; Herbert 1997), and the elements of the matrices  $A$ ,  $B$ ,  $C$ ,  $D$  can be found in Hein *et al.* (1994). They represent an initial-boundary-value problem that can be solved efficientl  $B, C, D$  can be found in Hein *et al.* (1994). They represent an initial-boundary-value problem that can be solved efficiently by a marching procedure in the *x*-direction with upstream effects neglected, an approach valid problem that can be solved efficiently by a marching procedure in the x-direction<br>with upstream effects neglected, an approach valid for convectively unstable flows<br>only. The partial differential equations system  $(3.4)$  with upstream effects neglected, an approach valid for convectively unstable flows<br>only. The partial differential equations system (3.4) is discretized by a first-order<br>backward Euler scheme in the x spatial direction and backward Euler scheme in the  $x$  spatial direction and fourth-order compact finite homogeneous Dirichlet boundary conditions at the wall  $(y = 0)$  and asymptotic differences in the *y* spatial direction. On the amplitude functions  $(\hat{u}, \hat{v}, \hat{w})$  we impose<br>homogeneous Dirichlet boundary conditions at the wall  $(y = 0)$  and asymptotic<br>boundary conditions enforcing exponential decay homogeneous Dirichlet boundary conditions at the wall  $(y = 0)$  and asymptotic<br>boundary conditions enforcing exponential decay at the upper boundary  $y_e$  (Hein *et*<br>*al.* 1994). Initial conditions are provided by a local pa boundary conditions enforcing exponential decay at the upper boundary  $y_e$  (Hein *et* al. 1994). Initial conditions are provided by a local parallel instability analysis using the same code, with the matrix D, all streamw al. 1994). Initial conditions are provided by a local parallel instability analysis using the same code, with the matrix D, all streamwise derivatives in matrices A, B and C and the wall-normal velocity component V of the the same code, with the matrix  $D$ , all streamwise derivatives in matrices  $A$ ,  $B$  and

C and the wall-normal velocity component V of the basic flow set to zero.<br>The spatial disturbance growth rate  $\sigma$  in local instability theory is described by<br>the negative imaginary part of the complex wavenumber  $\alpha$ , i. The spatial disturbance growth rate  $\sigma$  in local instability theory is described by<br>the negative imaginary part of the complex wavenumber  $\alpha$ , i.e.  $\sigma = -\alpha_i$ , whereas<br>in non-local theory there is an additional contribut the negative imaginary part of the complex wavenumber  $\alpha$ , i.e.  $\sigma = -\alpha_i$ , when non-local theory there is an additional contribution from the amplitude funct and various reasonable alternative definitions are possible; h and various reasonable alternative definitions are possible; here we use either

$$
\sigma_E = -\alpha_i + \frac{\partial}{\partial x} \ln(\sqrt{E}), \quad \text{with } E = \int_0^{y_e} (\|\hat{u}\|^2 + \|\hat{v}\|^2 + \|\hat{w}\|^2) \, \mathrm{d}y,\tag{3.5}
$$

or

**MATHEMATICAL,<br>PHYSICAL**<br>& ENGINEERING<br>SCIENCES

THE ROYAL

PHILOSOPHICAL<br>TRANSACTIONS

$$
\sigma_u = -\alpha_{\rm i} + Re\left(\frac{1}{\hat{u}_{\rm m}}\frac{\partial \hat{u}_{\rm m}}{\partial x}\right),\tag{3.6}
$$

where  $\hat{u}_m$  denotes the value of  $\hat{u}$  at the wall-normal coordinate where  $\|\hat{u}\|$  reaches its maximum.

#### (*b*) *The partial derivative eigenvalue problem*

The assumptions of slow growth of  $Q_{\rm b}$  and  $Q_{\rm p}$ , inherent in the PSE, may be The assumptions of slow growth of  $Q_{\rm b}$  and  $Q_{\rm p}$ , inherent in the PSE, may be<br>relaxed by fully resolving both the streamwise and the wall-normal spatial directions;<br>on account of this property we term the PEPS a *a* The assumptions of slow growth of  $Q_b$  and  $Q_p$ , inherent in the PSE, may be relaxed by fully resolving both the streamwise and the wall-normal spatial directions; on account of this property, we term the PEPS a *global* relaxed by fully resolving both the streamwise and the wall-normal spatial directions;<br>on account of this property, we term the PEPS a *global* instability analysis. In the<br>temporal framework, as considered here, the comp on account of this property, we term the PEPS a *global* instability analysis. In the temporal framework, as considered here, the complex function  $\Theta$  becomes simply

$$
\Theta = \beta z - \Omega t,\tag{3.7}
$$

 $\Theta = \beta z - \Omega t,$  (3.7)<br>where  $\beta$  is a real wavenumber parameter and  $\Omega$  is the sought complex eigenvalue<br>whose real part indicates frequency and whose positive imaginary part is the growth where  $\beta$  is a real wavenumber parameter and  $\Omega$  is the sought complex eigenvalue<br>whose real part indicates frequency and whose positive imaginary part is the growth<br>rate in time t of global disturbances. This definitio where  $\beta$  is a real wavenumber parameter and  $\Omega$  is the sought complex eigenvalue<br>whose real part indicates frequency and whose positive imaginary part is the growth<br>rate in time t of global disturbances. This definitio whose real part indicates frequency and whose positive imaginary part is the growth<br>rate in time t of global disturbances. This definition of  $\Theta$  suggests the second essential<br>difference between the PSE and the PEPS appr rate in time t of global disturbances. This definition of  $\Theta$  suggests the second essential difference between the PSE and the PEPS approaches, namely that any structure that an eigenmode may have in the streamwise direc difference between the PSE and the PEPS approaches, namely that any that an eigenmode may have in the streamwise direction  $x$ , including that clike disturbance considered by the PSE, will be captured by  $Q_p$  in (3.1).<br>Nu at an eigenmode may have in the streamwise direction x, including that of a wave-<br>e disturbance considered by the PSE, will be captured by  $Q_p$  in (3.1).<br>Numerical aspects of the solution of the two-dimensional partial de

like disturbance considered by the PSE, will be captured by  $Q_p$  in (3.1).<br>Numerical aspects of the solution of the two-dimensional partial derivative eigenvalue problem are discussed by Theofilis (1998). Compared with th value problem are discussed by Theofilis (1998). Compared with that work, we have<br>*Phil. Trans. R. Soc. Lond.* A (2000)

<sup>3234</sup> *V. Theo¯lis,S.HeinandU.Dallmann*

implemented here a novel version of the algorithm that applies to flow problems in implemented here a novel version of the algorithm that applies to flow problems in which the wavenumber vector is perpendicular to the plane on which the basic flow is defined (Theofilis 2000). Specifically straightforward implemented here a novel version of the algorithm that applies to flow problems in which the wavenumber vector is perpendicular to the plane on which the basic flow is defined (Theofilis 2000). Specifically, straightforwa is defined (Theofilis 2000). Specifically, straightforward redefinitions of the spanwise disturbance velocity component  $\hat{w}$  and the eigenvalue  $\Omega$  by i $\hat{w}$  and i $\Omega$ , respectively, is defined (Theofilis 2000). Specifically, straightforward redefinitions of the spanwise<br>disturbance velocity component  $\hat{w}$  and the eigenvalue  $\Omega$  by  $i\hat{w}$  and  $i\Omega$ , respectively,<br>result in a disturbance flow sys disturbance velocity component  $\hat{w}$  and the eigenvalue  $\Omega$  by  $i\hat{w}$  and  $i\Omega$ , respectively, result in a disturbance flow system with real coefficients that, compared with the full problem, requires approximately h result in a disturbance flow system with real coefficients that, compared with the full problem, requires approximately half the storage and runtime to solve. The real non-symmetric generalized eigenvalue problem for the full problem, requires approx<br>non-symmetric generalized  $\epsilon$ <br>which results is as follows: hows:<br>  $[\mathcal{L} - (\mathcal{D}_x U)]\hat{u} - (\mathcal{D}_y U)\hat{v} - \mathcal{D}_x \hat{p} = \Omega \hat{u},$  (3.8)<br>  $-(\mathcal{D}_x V)\hat{u} + [\mathcal{L} - (\mathcal{D}_y V)]\hat{v} - \mathcal{D}_y \hat{p} = \Omega \hat{v},$  (3.9)

$$
[\mathcal{L} - (\mathcal{D}_x U)]\hat{u} - (\mathcal{D}_y U)\hat{v} - \mathcal{D}_x \hat{p} = \Omega \hat{u},\tag{3.8}
$$

$$
-(\mathcal{D}_x V)\hat{u} + [\mathcal{L} - (\mathcal{D}_y V)]\hat{v} - \mathcal{D}_y \hat{p} = \Omega \hat{v},\tag{3.9}
$$

$$
\mathcal{L}\hat{w} - \beta \hat{p} = \Omega \hat{w},\tag{3.10}
$$

$$
\mathcal{L}\hat{w} - \beta \hat{p} = \Omega \hat{w},
$$
\n
$$
\mathcal{L}\hat{w} - \beta \hat{p} = \Omega \hat{w},
$$
\n
$$
\mathcal{D}_x \hat{u} + \mathcal{D}_y \hat{v} - \beta \hat{w} = 0,
$$
\n(3.10)\n
$$
(3.11)
$$

where

$$
\mathcal{L} = \left(\frac{1}{Re}\right)(\mathcal{D}_x^2 + \mathcal{D}_y^2 - \beta^2) - U\mathcal{D}_x - V\mathcal{D}_y, \quad \mathcal{D}_x = \frac{\partial}{\partial x} \text{ and } \mathcal{D}_y = \frac{\partial}{\partial y}.
$$

 $\mathcal{L} = \left(\frac{E}{Re}\right)(\mathcal{D}_x^2 + \mathcal{D}_y^2 - \beta^2) - U\mathcal{D}_x - V\mathcal{D}_y, \quad \mathcal{D}_x = \frac{\partial}{\partial x}$  and  $\mathcal{D}_y = \frac{\partial}{\partial y}$ .<br>For its solution we use spectral collocation on a rectangular Cartesian grid. The boundary conditions imposed ar  $\begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$  or its solution we use spectral collocation on a rectangular Cartesian grid. The boundary conditions imposed are the straightforward viscous boundary conditions on all disturbance velocity For its solution we use spectral collocation on a rectangular Cartesian grid. The boundary conditions imposed are the straightforward viscous boundary conditions on all disturbance velocity components at the wall and compa boundary conditions imposed are the straightforward viscous boundary conditions<br>on all disturbance velocity components at the wall and compatibility conditions for<br>the pressure; in the freestream the disturbance velocity c on all disturbance velocity components at the wall and compatibility conditions for<br>the pressure; in the freestream the disturbance velocity components and disturbance<br>pressure are required to decay. In an attempt to precl the pressure; in the freestream the disturbance velocity components and disturbance<br>pressure are required to decay. In an attempt to preclude disturbances from entering<br>the recirculating flow area from upstream, homogeneou pressure are required to decay. In an attempt to preclude disturbances from entering<br>the recirculating flow area from upstream, homogeneous Dirichlet boundary condi-<br>tions are imposed at the inflow. At the outflow boundary tions are imposed at the inflow. At the outflow boundary, linear extrapolation of all disturbance quantities from the interior of the integration domain is performed.

#### 4. Results

#### (*a*) *Short chordwise-wavelength instabilities studied by the PSE*

(a) Short chordwise-wavelength instabilities studied by the  $PSE$ <br>A classic means of linear analysis of short-wavelength instabilities in laminar separa-<br>tion bubbles is based on local theory in which the parallel flow ass A classic means of linear analysis of short-wavelength instabilities in laminar separation bubbles is based on local theory, in which the parallel flow assumption is made and the Orr–Sommerfeld equation (OSE) is solved (A A classic means of linear analysis of short-wavelength instabilities in laminar separation bubbles is based on local theory, in which the parallel flow assumption is made and the Orr-Sommerfeld equation (OSE) is solved (A tion bubbles is based on local theory, in which the parallel flow assumption is made<br>and the Orr–Sommerfeld equation (OSE) is solved (Allen  $\&$  Riley 1995). There is no<br>rational approach leading to the latter equation th and the Orr-Sommerfeld equation (OSE) is solved (Allen & Riley 1995). There is no rational approach leading to the latter equation that ignores both upstream influence<br>and downstream development of the basic flow; nevertheless, both the OSE and spa-<br>tial DNS predict, in close agreement, an explosive gr and downstream development of the basic flow; nevertheless, both the OSE and spa-<br>tial DNS predict, in close agreement, an explosive growth of convective instabilities<br>entering the laminar separated flow (Bestek *et al.* 1 tial DNS predict, in close agreement, an explosive growth of convective instabilities<br>entering the laminar separated flow (Bestek *et al.* 1989). On the other hand, an insta-<br>bility analysis approach which takes into acco entering the laminar separated flow (Bestek *et al.* 1989). On the other hand, an instability analysis approach which takes into account downstream development of the flow is based on the PSE. Due to the backflow within a flow is based on the PSE. Due to the backflow within a separation bubble one might a marching procedure, as used in non-local instability analyses, is not appropriate. suspect that there is a significant amount of information travelling upstream and<br>a marching procedure, as used in non-local instability analyses, is not appropriate.<br>However, there is strong indication from both experimen a marching procedure, as used in non-local instability analyses, is not appropriate.<br>However, there is strong indication from both experiment and DNS that thin lam-<br>inar separation bubbles are convectively unstable to sho However, there is strong indication from both experiment and DNS that thin laminar separation bubbles are convectively unstable to short-wavelength disturbances of small amplitude initiated upstream of the bubble (Dovgal % of small amplitude initiated upstream of the bubble (Dovgal *et al.* 1994; Rist *et al.* 1996). This encourages us to embark upon application of PSE for this type of flow.<br>As a first basic flow we take BF1 of Rist & Mau

the results of linear local theory with those of their DNS, the latter run at low

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THE ROYAL

PHILOSOPHICAL<br>TRANSACTIONS

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THE ROYAL

**PHILOSOPHICAL**<br>TRANSACTIONS



Figure 1. Comparison of non-local growth rates,  $\sigma_u$ , for BF1 obtained by NOLOT/PSE with<br>the DNS data of Rist & Maucher (1994) and our OSE solutions. Separation and reattachment<br>are denoted by 'S' and 'B' respectively. the DNS data of Rist & Maucher (1994) and our OSE solutions. Separation and reattachment are denoted by 'S' and 'R', respectively.

disturbance amplitudes; we use their growth rate data as a reference. Figure 1 shows disturbance amplitudes; we use their growth rate data as a reference. Figure 1 shows<br>the disturbance growth rates for  $f \approx 1719$  Hz (reduced frequency  $F = 1.8 \times 10^{-4}$ )<br>and three different (dimensional) spanwise wavenumbe ) disturbance amplitudes; we use their growth rate data as a reference. Figure 1 shows<br>the disturbance growth rates for  $f \approx 1719$  Hz (reduced frequency  $F = 1.8 \times 10^{-4}$ )<br>and three different (dimensional) spanwise wavenumbe the disturbance growth rates for  $f \approx 1719$  Hz (reduced frequency  $F = 1.8 \times 10^{-4}$ )<br>and three different (dimensional) spanwise wavenumbers  $\beta = 0$ , 400 and 800 m<sup>-1</sup>.<br>The corresponding wave angles  $\phi$  at separation point and three different (dimensional) spanwise wavenumbers  $\beta = 0$ , 400 and 800 m<sup>-1</sup>.<br>The corresponding wave angles  $\phi$  at separation point  $x_S$  are  $\phi \approx 0$ , 22 and 41°. The<br>linear local (OSE) and the non-local (PSE) resul The corresponding wave angles  $\phi$  at separation point  $x_S$  are  $\phi \approx 0$ , 22 and 41°. The linear local (OSE) and the non-local (PSE) results are compared with those obtained in the DNS of Rist & Maucher (1994). The non-lo linear local (OSE) and the non-local (PSE) results are compared with those obtained

The result of major significance in this figure is the agreement between the linear PSE and DNS results, which is found to be influenced by the wave angle. For two-The result of major significance in this figure is the agreement between the linear PSE and DNS results, which is found to be influenced by the wave angle. For two-<br>dimensional waves PSE delivers instability results that PSE and DNS results, which is found to be influenced by the wave angle. For two-<br>dimensional waves PSE delivers instability results that are in excellent agreement<br>with those obtained by DNS at all x positions, with the o dimensional waves PSE delivers instability results that are in excellent agreement<br>with those obtained by DNS at all  $x$  positions, with the obvious advantage that PSE<br>results are obtained at orders-of-magnitude lower com with those obtained by DNS at all  $x$  positions, with the obvious advantage that PSE results are obtained at orders-of-magnitude lower computing effort compared with DNS. The agreement is still very convincing for three-d results are obtained at orders-of-magnitude lower computing effort compared with<br>DNS. The agreement is still very convincing for three-dimensional waves, although<br>a discrepancy between PSE and DNS results that increases wi DNS. The agreement is still very convincing for three-dimensional waves, although<br>a discrepancy between PSE and DNS results that increases with wave angle is to<br>be found upstream of the bubble. There are at least three pos a discrepancy between PSE and DNS results that increases with wave angle is to<br>be found upstream of the bubble. There are at least three possible explanations for<br>this wave-angle dependent discrepancy. First, it might be t be found upstream of the bubble. There are at least three possible explanations for<br>this wave-angle dependent discrepancy. First, it might be thought that the upstream<br>influence neglected in PSE might cause PSE results to this wave-angle dependent discrepancy. First, it might be thought that the upstream<br>influence neglected in PSE might cause PSE results to deviate from those of the<br>DNS. Second, violation of the PSE assumption of a weakly n influence neglected in PSE might cause PSE results to deviate from those of the DNS. Second, violation of the PSE assumption of a weakly non-parallel flow in the neighbourhood of the separation point may be considered to b DNS. Second, violation of the PSE assumption of a weakly non-parallel flow in the neighbourhood of the separation point may be considered to be responsible for the discrepancy. Third, transients in the DNS data may cause t neighbourhood of the separation point may be considered to be responsible for the discrepancy. Third, transients in the DNS data may cause the small differences. The first reason cannot explain the discrepancies because these are present upstream of separation, in a region of attached boundary-layer flo first reason cannot explain the discrepancies because these are present upstream<br>of separation, in a region of attached boundary-layer flow where upstream effects<br>are indeed negligible but disappear in the second half of t of separation, in a region of attached boundary-layer flow where upstream effects<br>are indeed negligible but disappear in the second half of the bubble, despite the<br>existing backflow. In order for the second reason to be ve existing backflow. In order for the second reason to be verified, one should refer to *Phil. Trans. R. Soc. Lond.* A (2000)

<sup>3236</sup> *V. Theo¯lis,S.HeinandU.Dallmann*



the 2. Velocity profiles of the laminar basic flow BF1 and the amplitude functions c two-dimensional wave of  $f = 1719$  Hz at  $x = 1.984$  within the recirculating flow. two-dimensional wave of  $f = 1719$  Hz at  $x = 1.984$  within the recirculating flow.<br>the magnitude of the non-parallel terms in the attached, decelerating part of the flow

domain compared with that of the non-parallel terms in the recirculating zone.

We believe that the third reason plays an essential role, since disturbances are domain compared with that of the non-parallel terms in the recirculating zone.<br>We believe that the third reason plays an essential role, since disturbances are<br>generated in the DNS by periodic blowing and suction at the wa We believe that the third reason plays an essential role, since disturbances are generated in the DNS by periodic blowing and suction at the wall, immediately upstream of the region where flow deceleration is enforced. It generated in the DNS by periodic blowing and suction at the wall, immediately<br>upstream of the region where flow deceleration is enforced. It therefore takes some<br>downstream distance until these artificially introduced dist upstream of the region where flow deceleration is enforced. It therefore takes some downstream distance until these artificially introduced disturbances are converted into TS-like instabilities whose growth rate is no long downstream distance until these artificially introduced disturbances are converted<br>into TS-like instabilities whose growth rate is no longer affected by the actual means<br>of their generation. On the other hand, memory effec into TS-like instabilities whose growth rate is no longer affected by the actual means<br>of their generation. On the other hand, memory effects are shorter in our PSE anal-<br>yses, which were started with TS waves (solutions of their generation. On the other hand, memory effects are shorter in our PSE anal-<br>yses, which were started with TS waves (solutions of the OSE equation) imposed<br>at the inflow boundary; the distance from the end of the s yses, which were started with TS waves (solutions of the OSE equation) imposed<br>at the inflow boundary; the distance from the end of the suction strip to  $x = 1.5$ <br>corresponds to less than six disturbance wavefronts. It is at the inflow boundary; the distance from the end of the suction strip to  $x = 1.5$  corresponds to less than six disturbance wavefronts. It is well known that both the effects of flow non-parallelism and the effects of dis corresponds to less than six disturbance wavefronts. It is well known that both the effects of flow non-parallelism and the effects of disturbance history increase with disturbance wave angle. If, as we think, the differe effects of flow non-parallelism and the effects of disturbance history increase with disturbance wave angle. If, as we think, the differences are indeed caused by remaining transients, they should be more pronounced at sm disturbance wave angle. If, as we think, the differences are indeed caused by remaining transients, they should be more pronounced at smaller values of x. However, this could not be verified, because DNS growth-rate data ing transients, they should be more pronounced at smaller values of  $x$ . However, this could not be verified, because DNS growth-rate data for  $x < 1.5$  are not available from Rist  $\&$  Maucher (1994). These authors also p could not be verified, because DNS growth-rate data for  $x < 1.5$  are not available<br>from Rist & Maucher (1994). These authors also performed an OSE analysis (not<br>presented here), the results of which compare very well with presented here), the results of which compare very well with our linear local analysis (also shown in figure 1). The agreement between our OSE growth rate results presented here), the results of which compare very well with our linear local analysis (also shown in figure 1). The agreement between our OSE growth rate results and DNS is quite close, in line with the findings of Bestek ysis (also shown in figure 1). The agreement between our OSE growth rate results<br>and DNS is quite close, in line with the findings of Bestek *et al.* (1989) and others.<br>However, the agreement between OSE and DNS can be se and DNS is quite close, in line with the However, the agreement between OSE<br>that between PSE and DNS results.<br>Note that the largest growth rates However, the agreement between OSE and DNS can be seen to be much inferior to that between PSE and DNS results.<br>Note that the largest growth rates are of the same order of magnitude as the dis-

turbance wavenumbers, which means that the disturbances grow much more rapidly

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THE ROYAL



Figure 3. Isolines of constant non-local growth rate  $\sigma_E$  for BF1 (*a*) and BF2 (*b*). Frequencies f of unstable waves against scaled downstream distance are plotted. The outermost curves correspond to neutral stability Figure 3. Isolines of constant non-local growth rate  $\sigma_E$  for BF1 (*a*) and BF2 (*b*). Frequencies f of unstable waves against scaled downstream distance are plotted. The outermost curves correspond to neutral stability correspond to neutral stability and the increments in contour lines are  $\delta \sigma_E = 25$  and 10 for BF1 and BF2, respectively.

and BF2, respectively.<br>than those in Blasius flow. The non-local amplitude functions of the two-dimensional<br>wave of  $f \approx 1719$  Hz are plotted in figure 2 for the downstream position  $x = 1.984$ than those in Blasius flow. The non-local amplitude functions of the two-dimensional wave of  $f \approx 1719$  Hz are plotted in figure 2 for the downstream position  $x = 1.984$ , alongside those of the corresponding velocity prof wave of  $f \approx 1719$  Hz are plotted in figure 2 for the downstream position  $x = 1.984$ , alongside those of the corresponding velocity profiles of the basic flow.  $\hat{u}$  has maxima wave of  $f \approx 1719$  Hz are plotted in figure 2 for the downstream position  $x = 1.984$ , alongside those of the corresponding velocity profiles of the basic flow.  $\hat{u}$  has maximal inside the backflow region close to the w alongside those of the corresponding velocity profiles of the basic flow.  $\hat{u}$  has maxima<br>inside the backflow region close to the wall, at the inflection point of the stream-<br>wise velocity profile of the basic flow and inside the backflow region close to the wall, at the inflection point of the stream-<br>wise velocity profile of the basic flow and at the edge of the separated boundary<br>layer, while  $\hat{v}$  is rather large. Since the two-di wise velocity profile of the basic flow and at the edge of the separated boundary layer, while  $\hat{v}$  is rather large. Since the two-dimensional wave is the most amplified at almost all downstream positions (as seen in f layer, while  $\hat{v}$  is rather large. Since the two-dimensional wave is the most amplified<br>at almost all downstream positions (as seen in figure 1), we focus all subsequent<br>NOLOT/PSE studies on two-dimensional waves. The at almost all downstream positions (as seen in figure 1), we focus all subsequent<br>NOLOT/PSE studies on two-dimensional waves. The instability diagram for two-<br>dimensional short-wavelength disturbances is shown in figure 3. NOLOT/PSE studies on two-dimensional waves. The instability diagram for two-<br>dimensional short-wavelength disturbances is shown in figure 3. Isolines of constant<br>non-local growth rate (defined by 3.5) are plotted; the out dimensional short-wavelength disturbances is shown in figure 3. Isolines of constant<br>non-local growth rate (defined by 3.5) are plotted; the outermost isoline represents<br>the curve of neutral stability. This growth-rate def non-local growth rate (defined by 3.5) are plotted; the outermost isoline represents<br>the curve of neutral stability. This growth-rate definition was chosen because in the<br>separated flow region the disturbance amplitude fun the curve of neutral stability. This growth-rate definition was chosen because in the separated flow region the disturbance amplitude functions have several maxima in the wall-normal direction (as seen in figure 2). The in separated flow region the disturbance amplitude functions have several maxima in<br>the wall-normal direction (as seen in figure 2). The instability diagram for BF2 is<br>also shown in figure 3. Its general shape appears differe also shown in figure 3. Its general shape appears different from that of BF1, the latter having an instability region of the decelerating and separated flow embedded also shown in figure 3. Its general shape appears different from that of BF1, the latter having an instability region of the decelerating and separated flow embedded into the region of TS instability, unlike BF2 where the latter having an instabil<br>into the region of TS ins<br>also comments in  $\S 2$ ). (*b*) *Global disturbances addressed by the PEPS*

In the previous section we focused on instabilities of an oscillatory nature, the streamwise length-scale of which is small compared with that of the basic flow; In the previous section we focused on instabilities of an oscillatory nature, the streamwise length-scale of which is small compared with that of the basic flow; this structure is exploited by the ansatz  $(3.2)$ , and effi streamwise length-scale of which is small compared with that of the basic flow;<br>this structure is exploited by the ansatz (3.2), and efficient marching procedures<br>make a NOLOT/PSE approach the method of choice for accurate this structure is exploited by the ansatz  $(3.2)$ , and efficient marching procedures<br>make a NOLOT/PSE approach the method of choice for accurate results to be<br>delivered within reasonable computing time for this type of in make a NOLOT/PSE approach the method of choice for accurate results to be delivered within reasonable computing time for this type of instability. On the other hand, an analysis of disturbances developing over length-scal delivered within reasonable computing time for this type of instability. On the other<br>hand, an analysis of disturbances developing over length-scales comparable with<br>that of the basic flow using NOLOT/PSE may deliver inacc hand, an analysis of disturbances developing over length-scales comparable with that of the basic flow using  $\text{NOLOT/PSE}$  may deliver inaccurate results. Short of resorting to a spatial DNS in order to take both upstream resorting to a spatial DNS in order to take both upstream influence and downstream<br>*Phil. Trans. R. Soc. Lond.* A (2000)

THE ROYAL

**PHILOSOPHICAL**<br>TRANSACTIONS

**HYSICAL**<br>ENGINEERING<br>CIENCES **ATHEMATICAL** 

THE ROYAI

<sup>3238</sup> *V. Theo¯lis,S.HeinandU.Dallmann*



dow of the spectrum of global PE<br>in the neighbourhood of  $\Omega = 0$ .

in the neighbourhood of  $\Omega = 0$ .<br>development into account, disturbances having length-scales comparable with that of the basic flow are best analysed using numerical solutions of the partial derivative eigenvalue problem in which the available computing resources are devoted to the of the basic flow are best analysed using numerical solutions of the partial derivative<br>eigenvalue problem in which the available computing resources are devoted to the<br>simultaneous resolution of both the x and y spatial eigenvalue problem in which the available computing resources are devoted to the simultaneous resolution of both the x and y spatial directions. We concentrate here on presenting results for the BF2 basic flow, obtained u simultaneous resolution of both the x and y spatial directions. We concentrate here<br>on presenting results for the BF2 basic flow, obtained using the pressure-gradient<br>parameters  $\beta_0 = 100$  and  $\beta_1 = 300$  at  $Re = 10^6/48$ on presenting results for the BF2 basic flow, obtained using the pressure-gradient<br>parameters  $\beta_0 = 100$  and  $\beta_1 = 300$  at  $Re = 10^6/48$ . The eigenvalue spectrum in<br>the neighbourhood of  $\Omega = 0$  and for a spanwise wavenum parameters  $\beta_0 = 100$  and  $\beta_1 = 300$  at  $Re = 10^6/48$ . The eigenvalue spectrum in<br>the neighbourhood of  $\Omega = 0$  and for a spanwise wavenumber  $\beta = 20$  is shown in<br>figure 4. Stationary (Re{ $\Omega$ } = 0) as well as travelling the neighbourhood of  $\Omega = 0$  and for a spanwise wavenumber  $\beta = 20$  is shown in figure 4. Stationary (Re{ $\Omega$ } = 0) as well as travelling (Re{ $\Omega$ } ≠ 0) modes are to be found in this window of the spectrum, calculated by figure 4. Stationary ( $\text{Re}\{\Omega\} = 0$ ) as well as travelling ( $\text{Re}\{\Omega\} \neq 0$ ) modes are to be found in this window of the spectrum, calculated by the Arnoldi algorithm (Theofilis 1998). The travelling modes appear in sym preferential direction in z.<br>
1998). The travelling modes appear in symmetric pairs, indicating that there is no<br>
preferential direction in z.<br>
At this set of parameters, the most unstable mode is a stationary disturbance

preferential direction in z.<br>At this set of parameters, the most unstable mode is a stationary disturbance;<sup>†</sup> the eigenvector  $Q_p$  of this mode is presented in figure 5. In order to aid the discussion At this set of parameters, the most unstable mode is a stationary disturbance;<sup>†</sup> the eigenvector  $Q_p$  of this mode is presented in figure 5. In order to aid the discussion on this figure, the position of the laminar basi eigenvector  $\mathbf{Q}_{\rm p}$  of this mode is presented in figure 5. In order to aid the discussion<br>on this figure, the position of the laminar basic flow separation bubble is indicated<br>by a dotted line. From a numerical point on this figure, the position of the laminar basic flow separation bubble is indicated<br>by a dotted line. From a numerical point of view, we stress that our results show<br>that no effect of the downstream boundary conditions c by a dotted line. From a numerical point of view, we stress that our results show<br>that no effect of the downstream boundary conditions can be found in these PEPS<br>solutions, a result which we attribute to a novel treatment that no effect of the downstream boundary conditions can be found in these PEPS solutions, a result which we attribute to a novel treatment of the pressure bound-<br>ary conditions, which will be presented elsewhere. It suffices to mention here that<br>the same boundary conditions were shown to perform equa ary conditions, which will be presented elsewhere. It suffices to mention here that<br>the same boundary conditions were shown to perform equally well in another open-<br>system problem solved, that in the infinite swept attach the same boundary conditions were shown to perform equally well in another open-<br>system problem solved, that in the infinite swept attachment-line boundary layer<br>(Lin & Malik 1996; Theofilis 1997). Most of the activity in system problem solved, that in the infinite swept attachment-line boundary layer (Lin & Malik 1996; Theofilis 1997). Most of the activity in all disturbance eigenfunctions is confined within the boundary layer, with the n (Lin & Malik 1996; Theofilis 1997). Most of the activity in all disturbance eigenfunctions is confined within the boundary layer, with the neighbourhood of the inflow region being innocuous, as imposed by the inflow bound region being innocuous, as imposed by the inflow boundary condition. Significantly,<br>† A standing-wave pattern results from the linear superposition of the symmetric pairs of travelling

modes.

**MATHEMATICAL,<br>PHYSICAL<br>& ENGINEERING<br>SCIENCES** 

THE ROYAL<br>SOCIETY

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THE ROYAL

**PHILOSOPHICAL**<br>TRANSACTIONS

*Unsteadiness andthree-dimensionalityin separation bubbles* <sup>3239</sup>



stationary global mode in figure 4. Contour levels to be read as outer-to-inner values, where Figure 5. Normalized disturbance velocity components and pressure distribution of the unstable<br>stationary global mode in figure 4. Contour levels to be read as outer-to-inner values, where<br>available. (a) Re{ $\hat{u}$ }; 0.1( stationary global mode in figure 4. Contour levels to be read as outer-to-inner values, where<br>available. (a) Re $\{\hat{u}\}$ ; 0.1(0.2)0.9 (solid),  $-0.1(-0.1) - 0.3$  (dashed). (b) Re $\{\hat{v}\}$ ;  $-0.2(-0.2) - 0.8$ <br>(solid), 0.2(0.2) available. (a) Re{ $\hat{u}$ }; 0.1(0.2)0.9 (solid),  $-0.1(-0.1) - 0.3$  (dashed). (b) Re{ $\hat{v}$ };  $-0.2(-0.2) - 0.8$ <br>(solid), 0.2(0.2)0.8 (dashed). (c) Re{ $\hat{w}$ }; 0.2(0.2)0.8 (solid),  $-0.2(-0.2) - 0.8$  (dashed). (d) Re{ $\hat{p}$ } (solid),  $0.2(0.2)0.8$  (dashed). (c) Re{ $\hat{w}$ };  $0.2(0.2)0.8$  (solid),  $-0.2(-0.2)-0.8$  (dashed). (d) Re{ $\hat{p}$ };  $0.2(0.1)0.6$  (solid),  $-0.2, -0.3, -0.4$  (dashed). The phase of this eigenvector  $\theta(x, y)$  is approximately mately constant:  $\theta \approx \pi$ . In all results the location of the laminar separation bubble is indicated<br>by the dotted line. Note the different y-scales in the different eigenvectors.

the neighbourhood of the basic laminar flow separation point is also unaffected, as is clearly demonstrated by the level of activity of all disturbance velocity components the neighbourhood of the basic laminar flow separation point is also unaffected, as is<br>clearly demonstrated by the level of activity of all disturbance velocity components<br>and pressure in that region. The peak of both the clearly demonstrated by the level of activity of all disturbance velocity components<br>and pressure in that region. The peak of both the chordwise and the wall-normal lin-<br>ear disturbance velocity components is to be found and pressure in that region. The peak of both the chordwise and the wall-normal linear disturbance velocity components is to be found just downstream of the maximum extent of the primary recirculation region,  $x \approx 0.1$ , w ear disturbance velocity components is to be found just downstream of the maximum<br>extent of the primary recirculation region,  $x \approx 0.1$ , with lower-level linear activity<br>continuing past the point of primary reattachment. extent of the primary recirculation region,  $x \approx 0.1$ , with lower-level linear activity<br>continuing past the point of primary reattachment. Interestingly,  $\hat{w}$  (which is the<br>only source of three dimensionality in this l continuing past the point of primary reattachment. Interestingly,  $\hat{w}$  (which is the only source of three dimensionality in this linear framework) is mostly distributed within the primary separation bubble and has a te only source of three dimensionality in this linear framework) is mostly distributed<br>within the primary separation bubble and has a tendency to split the latter into<br>two regions of fluid moving in opposite directions. Press within the primary separation bubble and has a tendency to split the latter into<br>two regions of fluid moving in opposite directions. Pressure also has an interesting<br>signature, with almost purely sinusoidal disturbances *g* signature, with almost purely sinusoidal disturbances *generated* within the primary laminar bubble and persisting well after the laminar basic flow has reattached.

*Phil. Trans. R. Soc. Lond.* A (2000)

<sup>3240</sup> *V. Theo¯lis,S.HeinandU.Dallmann*



Figure 6. Local topological changes of flow structures associated with the formation of separation Figure 6. Local topological changes of flow structures associated with the formation of separation<br>bubbles, and global topological change associated with the onset of vortex shedding behind a<br>blunt body Figure 6. Loca<br>bubbles, and<br>blunt body.



Figure 7. Two different conjectures  $((a)$  and  $(b))$  for topological changes of a separation<br>bubble's structure associated with the onset of vortex shedding wo different conjectures  $((a)$  and  $(b))$  for topological changes of a bubble's structure associated with the onset of vortex shedding.

# bubble's structure associated with the onset of vortex shedding.<br>5. Topological conjectures for structural separation-bubble instability

5. Topological conjectures for structural separation-bubble instability<br>A separation bubble appears as a region of recirculating flow between a separation<br>point and a point of flow attachment located at a wall (figure 6a)  $\alpha$  separation bubble appears as a region of recirculating flow between a separation<br>point and a point of flow attachment located at a wall (figure 6a).<sup>†</sup> These two points<br>form half-saddles in a cross-sectional plane th A separation bubble appears as a region of recirculating flow between a separation<br>point and a point of flow attachment located at a wall (figure  $6a$ ).<sup>†</sup> These two points<br>form half-saddles in a cross-sectional plane tha point and a point of flow attachment located at a wall (figure  $6a$ ).<sup>†</sup> These two points form half-saddles in a cross-sectional plane that is parallel to the streamlines. At the wall, the locus of these points creates se form half-saddles in a cross-sectional plane that is parallel to the streamlines. At<br>the wall, the locus of these points creates separation and reattachment lines, respec-<br>tively, where a direction reversal of the near-wal the wall, the locus of these points creates separation and reattachment lines, respectively, where a direction reversal of the near-wall flow occurs since the wall-shear stress changes its sign. The single streamline (stre stress changes its sign. The single streamline (streamsurface) that connects these<br> $\dagger$  Figures 6, 8, 10 and 11 are also available at [http://www.sm.go.dlr.de/sm-sm](http://www.sm.go.dlr.de/sm-sm_info/TRTinfo/gallery.htm)\_info/TRTinfo/

[gallery.htm.](http://www.sm.go.dlr.de/sm-sm_info/TRTinfo/gallery.htm)

*Unsteadiness andthree-dimensionalityin separation bubbles* <sup>3241</sup>



Figure 8. Generic wall-shear stress distributions associated with the formation of multi-Figure 8. Generic wall-shear stress distributions associated with the formation of multi-<br>ple-structured separation bubbles. The asterisks indicate degenerate critical points for local<br>topological changes Figure 8. Generic w<br>ple-structured separat<br>topological changes.

topological changes.<br>two 'critical' points (lines) and the wall itself define a boundary of the region of<br>separated recirculating fluid which is characterized additionally by a centre point. two 'critical' points (lines) and the wall itself define a boundary of the region of separated, recirculating fluid, which is characterized, additionally, by a centre point of a 'vortex' (Dallmann *et al.* 1997). Hence at two 'critical' points (lines) and the wall itself define a boundary of the region of separated, recirculating fluid, which is characterized, additionally, by a centre point of a 'vortex' (Dallmann *et al.* 1997). Hence, at separated, recirculating fluid, which is characterized, additionally, by a centre point<br>of a 'vortex' (Dallmann *et al.* 1997). Hence, at incipient separation (not covered<br>by our present investigation), the two half-saddl of a 'vortex' (Dallmann *et al.* 1997). Hence, at incipient separation (not covered<br>by our present investigation), the two half-saddles and the centre point are created<br>simultaneously at the wall, and a separation bubble by our present investigation), the two half-saddles and the centre point are created simultaneously at the wall, and a separation bubble and a 'vortex' are created via *local* structural (topological) changes of a flow. In simultaneously at the wall, and a separation bubble and a 'vortex' are created via<br>local structural (topological) changes of a flow. In the structurally stable regime, the<br>simple bubble changes shape and grows in size but *local* structural (topological) changes of a flow. In the structurally stable regime, the simple bubble changes shape and grows in size but does not change its streamline topology. However, the topological structure of th simple bubble changes shape and grows in size but does not change its streamline<br>topology. However, the topological structure of the pressure field will, in general,<br>change at a different set of critical parameters, and th topology. However, the topological structure of the pressure field will, in general, change at a different set of critical parameters, and this in turn can initiate a change in the bubble structure. A more complicated, but change at a different set of critical parameters, and this in turn can initiate a change<br>in the bubble structure. A more complicated, but nevertheless local, structural flow<br>change occurs at the onset of flow separation b change occurs at the onset of flow separation behind a blunt body (figure  $6b$ ). In twochange occurs at the onset of flow separation behind a blunt body (figure  $6b$ ). In two-<br>dimensional flows around blunt bodies it is well known that the onset of vortex shed-<br>ding is associated with a break-up of saddle c dimensional flows around blunt bodies it is well known that the onset of vortex shedding is associated with a break-up of saddle connections. How can a simple bubble become unsteady? One conjecture might be that due to a h ding is associated with a break-up of saddle connections. How can a simple bubble<br>become unsteady? One conjecture might be that due to a hydro/aerodynamic insta-<br>bility a region of recirculating fluid might be instantaneou become unsteady? One conjecture might be that due to a hydro/aerodynamic instability a region of recirculating fluid might be instantaneously swept downstream in a way sketched in figure 7a, such that the points of separat bility a region of recirculating fluid might be instantaneously swept downstream in a<br>way sketched in figure  $7a$ , such that the points of separation and attachment merge<br>and release a patch of circulating fluid to convec way sketched in figure 7*a*, such that the points of separation and attachment merge<br>and release a patch of circulating fluid to convect downstream, while a new bubble<br>forms at the wall. Only local structural flow changes and release a patch of circulating fluid to convect downstream, while a new bubble<br>forms at the wall. Only local structural flow changes are associated with this type of<br>bubble unsteadiness and vortex shedding. Dallmann forms at the wall. Only local structural flow changes are associated with this type of bubble unsteadiness and vortex shedding. Dallmann  $et al.$  (1995) provide numerical evidence for the correctness of an earlier topologica bubble unsteadiness and vortex shedding. Dallmann *et al.* (1995) provide numerical evidence for the correctness of an earlier topological conjecture, namely that, prior to unsteadiness and/or vortex shedding, multiple re evidence for the correctness of an earlier topological conjecture, namely that, prior<br>to unsteadiness and/or vortex shedding, multiple recirculation zones appear within<br>a separation bubble and finally lead to a global stru to unsteadiness and/or vortex shedding, multiple recirculation zones appear within<br>a separation bubble and finally lead to a global structural flow change with mul-<br>tiple structurally unstable saddle-to-saddle connections. a separation bubble and finally lead to a global structural flow change with multiple structurally unstable saddle-to-saddle connections. The separation line could then stay almost stationary despite the vortex shedding p tiple structurally unstable saddle-to-saddle connections. The separation line could<br>then stay almost stationary despite the vortex shedding present, as sketched in the<br>sequence of instantaneous patterns of figure 7b.<br>There

sequence of instantaneous patterns of figure 7b.<br>There are several routes leading from a simple generic bubble to a multiple-<br>structured one before unsteadiness and vortex shedding sets in. In the most simple<br>multiple-str There are several routes leading from a simple generic bubble to a multiple-structured one before unsteadiness and vortex shedding sets in. In the most simple multiple-structured bubble case with several recirculating flow *Phil. Trans. R. Soc. Lond.* A (2000)

THE ROYAL<br>SOCIETY

PHILOSOPHICAL<br>TRANSACTIONS p

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THE ROYAL



Figure 9. Wall-shear distribution due to the basic and different linear amounts of the disturbance<br>streamwise velocity component. Shown are the separation (S) and reattachment (R) points as<br>generated by the presence of th Figure 9. Wall-shear distribution due to the basic and different linear an streamwise velocity component. Shown are the separation (S) and reagenerated by the presence of the two-dimensional linear disturbance.

generated by the presence of the two-dimensional linear disturbance.<br>structure of the associated wall-shear stress will be of the kind sketched in figure 8.<br>The so-called singularity or degeneracy in the wall-shear stress structure of the associated wall-shear stress will be of the kind sketched in figure 8.<br>The so-called singularity or degeneracy in the wall-shear stress graph (which appears<br>where its magnitude together with first and, pos The so-called singularity or degeneracy in the wall-shear stress graph (which appears where its magnitude together with first and, possibly, higher streamwise derivatives The so-called singularity or degeneracy in the wall-shear stress graph (which appears<br>where its magnitude together with first and, possibly, higher streamwise derivatives<br>vanish) indicates a local degeneracy where a second where its magnitude together with first and, possibly, higher streamwise derivatives<br>vanish) indicates a local degeneracy where a secondary bubble may form. In figure 9<br>the wall-shear pattern set up by the linear superpos vanish) indicates a local degeneracy where a secondary bubble may form. In figure 9 the wall-shear pattern set up by the linear superposition of the basic flow and a linearly small amount of the streamwise disturbance vel the wall-shear pattern set up by the linear superposition of the basic flow and a linearly small amount of the streamwise disturbance velocity component  $\hat{u}$  shown in figure 5 is shown; the latter amount is indicated o Inearly small amount of the streamwise disturbance velocity component  $\hat{u}$  shown in figure 5 is shown; the latter amount is indicated on the figure in percentage terms.<br>The effect of the presence of the stationary unst figure 5 is shown; the latter amount is indicated on the figure in percentage terms.<br>The effect of the presence of the stationary unstable mode on the primary separation<br>and reattachment points may be seen in this figure. The effect of the presence of the stationary unstable mode on the primary separation<br>and reattachment points may be seen in this figure. While the separation line (in<br>three dimensions) remains unaffected in the presence of and reattachment points may be seen in this figure. While the separation line (in three dimensions) remains unaffected in the presence of a linearly unstable global mode, the reattachment point of the basic flow at suffici three dimensions) remains unaffected in the presence of a linearly unstable global<br>mode, the reattachment point of the basic flow at sufficiently high amplitudes of the<br>linearly unstable mode becomes a secondary separation mode, the reattachment point of the basic flow at sufficiently high amplitudes of the linearly unstable mode becomes a secondary separation point, while both upstream and downstream of this point, secondary reattachment is

and downstream of this point, secondary reattachment is generated *linearly*.<br>Let us now consider the onset of three dimensionality in a simple two-dimensional<br>separation bubble as sketched in figure 10.<sup>†</sup> The separation Let us now consider the onset of three dimensionality in a simple two-dimensional separation bubble as sketched in figure  $10.$ † The separation and attachment lines of the two-dimensional flow (upper and middle parts of f the two-dimensional flow (upper and middle parts of figure 10) will always be structurally unstable against any three-dimensional (steady or unsteady) perturbation.

<sup>y</sup> The mathematical framework of a description of the structural changes associated with local and <sup>†</sup> The mathematical framework of a description of the structural changes associated with local and (instantaneous) global topological flow changes and the set of 'elementary (three-dimensional) topological flow structures † The mathematical framework of a description of the structural changes associate<br>(instantaneous) global topological flow changes and the set of 'elementary (three-dimension)<br>flow structures' which can appear at a rigid wa *Phil. Trans. R. Soc. Lond.* A (2000)

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THE ROYAI

**PHILOSOPHICAL**<br>TRANSACTIONS

**MATHEMATICAL,<br>PHYSICAL<br>& ENGINEERING<br>SCIENCES** 

THE ROYAL<br>SOCIETY

**MATHEMATICAL,<br>PHYSICAL<br>& ENGINEERING<br>SCIENCES** 

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**PHILOSOPHICAL**<br>TRANSACTIONS

**MATHEMATICAL,<br>PHYSICAL<br>& ENGINEERING<br>SCIENCES** 

THE ROYAL<br>SOCIETY

**PHILOSOPHICAL**<br>TRANSACTIONS ŏ

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*Unsteadiness andthree-dimensionalityin separation bubbles* <sup>3243</sup>



Figure 10. Streamline topology of a simple generic separation bubble (upper). Wall-flow<br>pattern for a two-dimensional simple separation bubble (middle). Wall-flow pattern for a Figure 10. Streamline topology of a simple generic separation bubble (upper). Wall-flow pattern for a two-dimensional simple separation bubble (middle). Wall-flow pattern for a three-dimensional separation 'bubble' (lower pattern for a two-dimensional simple separation bubble (middle). Wall-flow pattern for a three-dimensional separation 'bubble' (lower).



Figure 11. Streamline topology of a multiple-structured separation bubble (upper). Wall-flow 11. Streamline topology of a multiple-structured separation bubble (upper). Wall pattern for the three-dimensional multiple-structured separation bubble (lower).

<sup>3244</sup> *V. Theo¯lis,S.HeinandU.Dallmann*

**MATHEMATICAL,<br>PHYSICAL**<br>& ENGINEERING<br>SCIENCES

THE ROYAL

**PHILOSOPHICAL**<br>TRANSACTIONS

*AATHEMATICAL,<br>'HYSICAL<br>k ENGINEERING<br>CIENCES* 

The structurally stable configuration will be formed via a sequence of saddle and The structurally stable configuration will be formed via a sequence of saddle and nodal points (either nodes or foci) along the separation as well as the attachment  $\lim_{\epsilon \to 0} f(s)$ . The resulting wall-streamline pattern ( The structurally stable configuration will be formed via a sequence of saddle and<br>nodal points (either nodes or foci) along the separation as well as the attachment<br>line(s). The resulting wall-streamline pattern (lower par nodal points (either nodes or foci) along the separation as well as the attachment<br>line(s). The resulting wall-streamline pattern (lower part of figure 10) does not nec-<br>essarily differ much from the wall-flow pattern for line(s). The resulting wall-streamline pattern (lower part of figure 10) does not necessarily differ much from the wall-flow pattern for a two-dimensional flow; however, "three dimensionality' will always be observed close essarily differ much from the wall-flow pattern for a two-dimensional flow; however, "three dimensionality" will always be observed close to the critical points, i.e. close to points of separation and/or attachment. What w "three dimensionality" will always be observed close to the critical points, i.e. close to points of separation and/or attachment. What will be the effect of three-dimensional disturbances on the above-discussed incipient points of separation and/or attachment. What will be the effect of three-dimensional disturbances on the above-discussed incipient structural change from a simple two-<br>dimensional bubble into a multiple-structured one (wh disturbances on the above-discussed incipient structural change from a simple two-<br>dimensional bubble into a multiple-structured one (which may appear and exist as<br>a steady flow structure prior to flow unsteadiness and vor dimensional bubble into a multiple-structured one (which may appear and exist as<br>a steady flow structure prior to flow unsteadiness and vortex shedding!)? For sim-<br>plicity let us assume that the disturbance flow is periodi plicity let us assume that the disturbance flow is periodic in the spanwise direction (as assumed in the PEPS and NOLOT/PSE analyses). The additional amplification of a wall-shear stress of the disturbance flow will increa plicity let us assume that the disturbance flow is periodic in the spanwise direction (as assumed in the PEPS and NOLOT/PSE analyses). The additional amplification of a wall-shear stress of the disturbance flow will increa (as assumed in the PEPS and NOLOT/PSE analyses). The additional amplification of a wall-shear stress of the disturbance flow will increase and reduce the local mean-flow wall-shear stress in a spanwise-periodic way. The r mean-flow wall-shear stress in a spanwise-periodic way. The result may be found<br>in figure 8, where the singularities in the wall-shear stress distribution (along x) will be structurally stabilized with additional critical points of attachment and sepin figure 8, where the singularities in the wall-shear stress distribution (along  $x$ ) will be structurally stabilized with additional critical points of attachment and separation occurring at one spanwise location, while will be structurally stabilized with additional critical points of attachment and separation occurring at one spanwise location, while the simple bubble structure will be preserved at other spanwise locations. Hence, the aration occurring at one spanwise location, while the simple bubble structure will<br>be preserved at other spanwise-locations. Hence, the  $x$  spatial oscillatory wall-shear<br>stress behaviour (figure 8) of a spanwise-periodic be preserved at other spanwise locations. Hence, the  $x$  spatial oscillatory wall-shear<br>stress behaviour (figure 8) of a spanwise-periodic disturbance flow of sufficient ampli-<br>tude must immediately lead to a pronounced t stress behaviour (figure 8) of a spanwise-periodic disturbance flow of sufficient amplitude must immediately lead to a pronounced three-dimensional wall-flow pattern if a multiple-structured bubble occurs (upper part of fi tude must immediately lead to a pronounced three-dimensional wall-flow pattern if<br>a multiple-structured bubble occurs (upper part of figure 11). The above-presented<br>results of the PEPS instability analysis clearly support a multiple-structured bubble occurs (upper part of figure 11). The above-presented<br>results of the PEPS instability analysis clearly support these topological conjectures,<br>namely the growth of a stationary disturbance mode results of the PEPS instability analysis clearly support these topological conjectures,<br>namely the growth of a stationary disturbance mode leads to a multiple-structured<br>separation 'bubble' prior to flow unsteadiness and v namely the growth of a stationary disturbance mode leads to a multiple-structured<br>separation 'bubble' prior to flow unsteadiness and vortex shedding. However, such a<br>'bubble' breaks up into a structure with complex three-d separation 'bubble' prior to flow unsteadiness and vortex shedding. However, such a<br>
'bubble' breaks up into a structure with complex three-dimensional topology due to<br>
the inherent three-dimensional character of the stati bubble' breaks up into a structure with complex three-dimensional topology due to<br>the inherent three-dimensional character of the stationary spanwise-periodic distur-<br>bance mode. Three dimensionality will leave its footpri the inherent three-dimensional character of the stationary spanwise-periodic distur-<br>bance mode. Three dimensionality will leave its footprints in the wall-flow pattern<br>especially within the 'reattachment zone', i.e. where

### 6. Discussion and outlook

6. Discussion and outlook<br>In this paper we subject a steady laminar two-dimensional boundary-layer flow that<br>incorporates a recirculation zone to two distinct linear instability analyses and report. In this paper we subject a steady laminar two-dimensional boundary-layer flow that<br>incorporates a recirculation zone to two distinct linear instability analyses and report<br>two sets of significant findings. First, we apply In this paper we subject a steady laminar two-dimensional boundary-layer flow that<br>incorporates a recirculation zone to two distinct linear instability analyses and report<br>two sets of significant findings. First, we apply incorporates a recirculation zone to two distinct linear instability analyses and report<br>two sets of significant findings. First, we apply PSE in order to study instability of<br>a wall-bounded flow that encompasses a separat two sets of significant findings. First, we apply PSE in order to study instability of<br>a wall-bounded flow that encompasses a separation bubble. The agreement between<br>PSE and spatial DNS results ranges from very good for a wall-bounded flow that encompasses a separation bubble. The agreement between PSE and spatial DNS results ranges from very good for three-dimensional disturbances to excellent for plane instability waves. Pauley *et al.* dimensional DNS and reported that the Strouhal number  $St = f \delta_{2,S}/U_e$  based on bances to excellent for plane instability waves. Pauley *et al.* (1990) performed two-<br>dimensional DNS and reported that the Strouhal number  $St = f\delta_{2,S}/U_e$  based on<br>the vortex-shedding frequency f, the momentum thickness dimensional DNS and reported that the Strouhal number  $St = f\delta_{2,S}/U_e$  based on<br>the vortex-shedding frequency f, the momentum thickness  $\delta_2$  and the boundary-<br>layer edge velocity  $U_e$  at separation is  $St = 0.00686 \pm 0.6\%$ , the vortex-shedding frequency f, the momentum thickness  $\delta_2$  and the boundary-<br>layer edge velocity  $U_e$  at separation is  $St = 0.00686 \pm 0.6\%$ , and is independent of<br>Reynolds number and pressure gradient; this result cor layer edge velocity  $U_e$  at separation is  $St = 0.006 86 \pm 0.6\%$ , and is independent of Reynolds number and pressure gradient; this result corresponds to dimensional frequencies of  $f = 1180$  Hz and  $f = 986$  Hz for BF1 and Reynolds number and pressure gradient; this result corresponds to dimensional frequencies of  $f = 1180$  Hz and  $f = 986$  Hz for BF1 and BF2, respectively. Both match the most unstable frequencies found in our non-local inst quencies of  $f = 1180$  Hz and  $f = 986$  Hz for BF1 and BF2, respectively. Both match<br>the most unstable frequencies found in our non-local instability analyses very well, as<br>can be seen in figure 3. This finding is in line w the most unstable frequencies found in our non-local instability analyses very well, as<br>can be seen in figure 3. This finding is in line with those of Bestek *et al.* (1989), who<br>reported that the mechanism of unsteadines can be seen in figure 3. This finding is in line with those of Bestek *et al.* (1989), who reported that the mechanism of unsteadiness of separation bubbles is closely linked reported that the mechanism of unsteadiness of separation bubbles is closely linked<br>to the growth of disturbances travelling through the bubble. The agreement between<br>PSE and DNS indicates that the upstream influence withi to the growth of disturbances travelling through the bubble. The agreement between<br>PSE and DNS indicates that the upstream influence within a laminar separation<br>bubble is indeed negligible for the short-wavelength disturba PSE and DNS indicates that the upstream influence within a laminar separation bubble is indeed negligible for the short-wavelength disturbances considered, as long as the prevailing instabilities are of a convective nature

*Phil. Trans. R. Soc. Lond.* A (2000)

*Unsteadiness andthree-dimensionalityin separation bubbles* <sup>3245</sup>

large disturbance growth rates in flows with laminar separation renders nonlinear PSE analyses much more challenging than in attached boundary layers.

The second significant finding presented in this paper concerns the existence of global instability modes inaccessible to local analyses. Solutions of the partial deriva-The second significant finding presented in this paper concerns the existence of global instability modes inaccessible to local analyses. Solutions of the partial derivative eigenvalue problem have been discovered that cor global instability modes inaccessible to local analyses. Solutions of the partial deriva-<br>tive eigenvalue problem have been discovered that correspond to both stationary and<br>travelling global instabilities, periodic in the tive eigenvalue problem have been discovered that correspond to both stationary and<br>travelling global instabilities, periodic in the spanwise z-direction. While we present<br>solutions at conditions supporting linear growth o travelling global instabilities, periodic in the spanwise  $z$ -direction. While we present solutions at conditions supporting linear growth of one stationary mode alone, results not presented here indicate that both travel solutions at conditions supporting linear growth of one stationary mode alone, results<br>not presented here indicate that both travelling and other stationary modes may<br>become linearly unstable at different parameters. From not presented here indicate that both travelling and other stationary modes may<br>become linearly unstable at different parameters. From the point of view of mainte-<br>nance of laminar flow by instability control, the existenc become linearly unstable at different parameters. From the point of view of mainte-<br>nance of laminar flow by instability control, the existence of the new modes suggests<br>that engineering methods of control of small-amplitu nance of laminar flow by instability control, the existence of the new modes suggests<br>that engineering methods of control of small-amplitude disturbances that aim at the<br>frequencies of TS-like disturbances—as delivered by that engineering methods of control of small-amplitude disturbances that aim at the frequencies of TS-like disturbances—as delivered by DNS or PSE and well approximated by an OSE solution—will have no impact on the global frequencies of TS-like disturbances—as delivered by DNS or PSE and well approxi-<br>mated by an OSE solution—will have no impact on the global modes; the frequencies<br>of the latter also have to be identified. However, a contro mated by an OSE solution—will have no impact on the global modes; the frequencies of the latter also have to be identified. However, a control method based on frequency information alone is bound to fail altogether if the of the latter also have to be identified. However, a control method based on frequency information alone is bound to fail altogether if the unstable global mode is a stationary disturbance. A further significant result rel tionary disturbance. A further significant result related to the new global modes tionary disturbance. A further significant result related to the new global modes<br>concerns past conjectures that used topological arguments in an attempt to explain<br>the origins of unsteadiness and three dimensionality of l concerns past conjectures that used topological arguments in an attempt to explain<br>the origins of unsteadiness and three dimensionality of laminar separated flow. A<br>linear mechanism is found and presented which, on account the origins of unsteadiness and three dimensionality of laminar separated flow. A linear mechanism is found and presented which, on account of the instability of the global mode and its spanwise periodicity, respectively, linear mechanism is found and presented which, on account of the instability of the global mode and its spanwise periodicity, respectively, may lead to unsteadiness and three dimensionality in line with a multitude of experimental observations (see, for example, Dovgal *et al.* 1994). Being solutions of the equations of motion, the global modes should be reproducible by spatial DNS; on t example, Dovgal *et al.* 1994). Being solutions of the equations of motion, the global modes should be reproducible by spatial DNS; on the other hand, the wave-like

identifying the PEPS modes; both questions are currently under investigation.<br>This paper is devoted to the memory of our colleague and friend Horst Bestek. Discussions<br>Ulrich Rist as well as his DNS data are appreciated. This paper is devoted to the memory of our colleague and friend Horst Bestek. Discussions with

#### References

- **Allen, T. & Riley, N. 1995 Absolute and convective instabilities in separation bubbles.** *[Aero. J.](http://matilde.ingentaselect.com/nw=1/rpsv/cgi-bin/linker?ext=a&reqidx=/0001-9240^28^2999L.439[aid=541363,csa=0001-9240^26vol=99^26iss=990^26firstpage=439])* **99**  $A39-448$ len, T. & Riley<br>**99**, 439–448.<br>rtolotti F. P Allen,T. & Riley, N. 1995 Absolute and convective instabilities in separation bubbles. Aero. J.<br>
99, 439–448.<br>
Bertolotti, F. P., Herbert, Th. & Spalart, P. R. 1992 Linear and nonlinear stability of the Blasius<br>
boundary
- **99**, 439–448.<br>rtolotti, F. P., Herbert, Th. & Spalart, P. R. 199:<br>boundary layer. *J. Fluid Mech.* **242**, 441–474.<br>stek. H. Gruber. K. & Essel. H. 1989 Self-excite. Bertolotti, F. P., Herbert, Th. & Spalart, P. R. 1992 Linear and nonlinear stability of the Blasius<br>boundary layer. J. Fluid Mech. 242, 441–474.<br>Bestek, H., Gruber, K. & Fasel, H. 1989 Self-excited unsteadiness of laminar
- boundary layer. *J. Fluid Mech.* **242**, 441–474.<br>stek, H., Gruber, K. & Fasel, H. 1989 Self-excited unsteadiness of laminar separation bubbles<br>caused by natural transition. In *Proc. Conf. on the Prediction and Exploitatio Flows, H.*, Gruber, K. & Fasel, H. 1989 Self-excited unsteadiness<br>caused by natural transition. In *Proc. Conf. on the Prediction*<br>*Flows, 18-20 October 1989*. The Royal Aeronautical Society.<br>Jew W. B. 1971 A numerical st caused by natural transition. In *Proc. Conf. on the Prediction and Exploitation of Separated*<br>Flows, 18-20 October 1989. The Royal Aeronautical Society.<br>Briley, W. R. 1971 A numerical study of laminar separation bubbles
- Flows, 18-20 October 1989. The Royal Aeronautical Society.<br>Briley, W. R. 1971 A numerical study of laminar separation bubbles using the Navier-Stokes<br>equations. *J. Fluid Mech.* 47, 713–736.<br>Cebeci, T. & Stewartson, K. 198 Briley, W. R. 1971 A numerical study of laminar separation bubbles using the Navier-Stokes equations. *[J. Fluid Mech.](http://matilde.ingentaselect.com/nw=1/rpsv/cgi-bin/linker?ext=a&reqidx=/0022-1120^28^29133L.287[aid=541366,csa=0022-1120^26vol=133^26iss=^26firstpage=287])* 47, 713-736.<br>Cebeci, T. & Stewartson, K. 1983 On the calculation of separation bubbles. *J. Fluid Mech.*
- Cebeci,T. & Stewartson, K. 1983 On the calculation of separation bubbles. *J. Fluid Mech.* 133,<br>287–296.<br>Dallmann, U. 1988 Three-dimensional vortex structures and vorticity topology. In *Proc. IUTAM*<br>Sump on Fundamental A
- 287–296.<br>Illmann, U. 1988 Three-dimensional vortex structures and vorticity topology. In *F*<br>*Symp. on Fundamental Aspects of Vortex Motion, Tokyo, Japan*, pp. 183–189.<br>Illmann, U. Horborg, Th. Gobing, H. Su. W. H. & Zhang Symp. on Fundamental Aspects of Vortex Motion, Tokyo, Japan, pp. 183–189.<br>Dallmann, U., Herberg, Th., Gebing, H., Su, W.-H. & Zhang, H.-Q. 1995 Flow field diagnostics:
- topological flow changes and spatio-temporal flow structures. AIAA 33rd Aerospace Sciences Meeting and Exhibit, AIAA-95-0791. topological flow changes and spatio-temporal flow structures. AIAA 33rd Aerospace Sciences<br>Meeting and Exhibit, AIAA-95-0791.<br>Dallmann, U., Vollmers, H., Su, W.-H. & Zhang, H.-Q. 1997 Flow topology and tomography<br>for vorte
- Meeting and Exhibit, AIAA-95-0791.<br>illmann, U., Vollmers, H., Su, W.-H. & Zhang, H.-Q. 1997 Flow topology and tomography<br>for vortex identification in unsteady and in three-dimensional flows. In *Proc. IUTAM Symp.*<br>on *Simu on Simulation and Identification of Organized Structures in Flow topology and tomography*<br>for vortex identification in unsteady and in three-dimensional flows. In *Proc. IUTAM Symp.*<br>*on Simulation and Identification of O* for vortex id<br> *on Simulatic*<br> *May 1997. Phil. Trans. R. Soc. Lond.* A (2000)

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<sup>3246</sup> *V. Theo¯lis,S.HeinandU.Dallmann*

- $V.$  Theoplits, S. Hein and U. Dalimann<br>Dovgal, A. V., Kozlov, V. V. & Michalke, A. 1994 Laminar boundary layer separation: instability<br>and associated phenomena. *Prog. Aerosnace Sci*. 30, 61–94 begal, A. V., Kozlov, V. V. & Michalke, A. 1994 Laminar bound<br>and associated phenomena. *Prog. Aerospace Sci.* **30**, 61–94.<br>in S. Berteletti E. P. Simon M. Hanifi. A. & Henningson and associated phenomena. *Prog. Aerospace Sci.* **30**, 61–94.<br>Hein, S., Bertolotti, F. P., Simen, M., Hanifi, A. & Henningson, D. 1994 Linear nonlocal insta-
- bility analysis—the linear NOLOT code. Internal report DLR IB 223-94 A56. Hein, S., Bertolotti, F. P., Simen, M., Hanin, A. & Henningson, D. 1994 Linear nonlocal instability analysis—the linear NOLOT code. Internal report DLR IB 223-94 A56.<br>Hein, S., Theofilis, V. & Dallmann, U. 1998 Unsteadines
- bility analysis—the linear NOLOT code. Internal report DLR IB 223-94 A56.<br>in, S., Theofilis, V. & Dallmann, U. 1998 Unsteadiness and three-dimensionality of steady two-<br>dimensional laminar separation bubbles as result of l dimensional laminar separation bubbles as result of linear instability mechanisms. Internal report DLR IB 223-98 A39.
- Herbert, Th. 1997 Parabolized stability equations. *A. Rev. Fluid Mech.* **29**, 245–283.
- reportDLR IB 223-98 A39.<br>Herbert, Th. 1997 Parabolized stability equations. A. Rev. Fluid Mech. 29, 245–283.<br>Howarth, L. 1938 On the solution of the laminar boundary layer equations *Proc. R. Soc. Lond.*<br>A 164, 547–579. rbert, Th. 1997 Pa<br>warth, L. 1938 Or<br>A 164, 547–579.<br>A B S & Molik Lin, R.-S. & Malik, M. R. 1996 On the stability of the attachment-line boundary layers. Part 1.
- The incompressible swept Hiemenz flow. *J. Fluid Mech.* **311**, 239–255. Lin, K.-S. & Malik, M. K. 1996 On the stability of the attachment-line boundary layers. Part 1.<br>The incompressible swept Hiemenz flow. *J. Fluid Mech.* **311**, 239–255.<br>Pauley, L. L., Moin, P. & Reynolds, W. C. 1990 The st
- The incompressible swept Hiem<br>
uley, L. L., Moin, P. & Reynol<br> *J. Fluid Mech.* **220**, 397–411.<br>
<sup>++</sup> U. <sup>*V.*</sup> Mauchar, U. 1004 Di
- J. Fluid Mech. 220, 397–411.<br>Rist, U. & Maucher, U. 1994 Direct numerical simulation of 2-D and 3-D instability waves *J. Fluid Mech.* **220**, 397–411.<br>st, U. & Maucher, U. 1994 Direct numerical simulation of 2-D and 3-D instability waves<br>in a laminar separation bubble. In *AGARD-CP-551, Application of Direct and Large Eddy*<br>Simulation to *Simulation to Transition and Turbulence*, pp. 34-1-34-7.<br> *Simulation to Transition and Turbulence*, pp. 34-1-34-7.<br> *Simulation to Transition and Turbulence*, pp. 34-1-34-7. In a laminar separation bubble. In AGARD-CP-551, Application of Direct and Large Eddy<br>
Simulation to Transition and Turbulence, pp. 34-1–34-7.<br>
Rist, U., Maucher, U. & Wagner, S. 1996 Direct numerical simulation of some fu
- Simulation to Transition and Turbulence, pp. 34-1–34-7.<br>st, U., Maucher, U. & Wagner, S. 1996 Direct numerical simulation of some fundamental<br>problems related to transition in laminar separation bubbles. In *Proc. Computat Br.* U., Maucher, U. & Wagner, S. 1996 Direct<br>problems related to transition in laminar separa<br>*Dynamics Conf. ECCOMAS '96*, pp. 319–325.<br>plichting H. 1979 Boundary layer theory McGre problems related to transition in laminar separation bubbles. In *Proc. Computational Fluid Dynamics Conf. ECCOMAS* '96, pp. 319–325.<br>Schlichting, H. 1979 *Boundary-layer theory*. McGraw-Hill.

- Dynamics Conf. ECCOMAS 96, pp. 319–325.<br>Schlichting, H. 1979 Boundary-layer theory. McGraw-Hill.<br>Spalart, P. R. & Coleman, G. N. 1997 Numerical study of a separation bubble with heat transfer.<br>Eur. J. Mech. B.16, 169–189. *Alichting, H. 1979 Boundary-lay*<br> *Alart, P. R. & Coleman, G. N. 19*<br> *Eur. J. Mech.* B 16, 169–189.<br> **Alarge M. 1997 On the verifice** Eur. J. Mech. B  $16$ , 169–189.<br>Theofilis, V. 1997 On the verification and extension of the Görtler–Hämmerlin assumption in
- Eur. J. Mech. B 16, 169–189.<br>
eofilis, V. 1997 On the verification and extension of the Görtler–Hämmerlin assumption in<br>
three-dimensional incompressible swept attachment-line boundary layer flow. Internal report<br>
DLB IB eofilis, V. 1997 On the<br>three-dimensional inco<br>DLR IB 223-97 A44.<br>coofilis, V. 1008 Lincon three-dimensional incompressible swept attachment-line boundary layer flow. Internal report<br>DLR IB 223-97 A44.<br>Theofilis, V. 1998 Linear instability in two spatial dimensions. In *Proc. Computational Fluid*<br>Dimagnics Conf.
- *DLR IB 223-97 A44.*<br> *eofilis, V. 1998 Linear instability in two spatial dimensions. In Proc. Computational*<br> *Dynamics Conf. ECCOMAS '98, Athens, Greece, 7–11 September 1998*, pp. 547–552.<br>
reofilis V. 2000 Clobally unst Dynamics Conf. ECCOMAS '98, Athens, Greece,  $\gamma$ -11 September 1998, pp. 547–552.<br>Theofilis, V. 2000 Globally unstable flows in open cavities. AIAA paper no. 2000-1965.

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